

**SOME EXTENSIONS OF THE
ROOT LOCUS METHOD**

E. GRANT REES

Library
U. S. Naval Postgraduate School
Monterey, California

Library
U. S. Naval Postgraduate School
Monterey, California

SOME EXTENSIONS OF THE
ROOT LOCUS METHOD

by

E. Grant Rees
Lieutenant, United States Navy
B. S., United States Naval Academy, 1959



Submitted in partial fulfillment
for the degree of

MASTER OF SCIENCE IN ELECTRICAL ENGINEERING

from the

UNITED STATES NAVAL POSTGRADUATE SCHOOL
May 1966

17-513
P-10

ABSTRACT

Simple analytical methods for the determination of the angle of emergence from complex poles and zeros are developed. Several case studies of root loci emerging from complex zeros are analyzed to determine criterion for the loci starting from complex pole to end on complex zero, as a pre-requisite study for "cancel-compensation" and self-adaptive compensators.

The writer wishes to thank Dr. George J. Thaler for his advice and encouragement in preparing this paper.

TABLE OF CONTENTS

Chapter	Title	Page
I	Introduction	7
II	Angle of Emergence	11
	Determination of the Angle of Emergence	11
	Complex Residues and the Angle of Emergence	14
	Effect of Complex Zeros near Complex Poles	15
III	Root Loci Leaving Complex Poles	25
	Data Designation	25
	Observation from Data	52
IV	Conclusions	59
Appendix I	Computer Programs	63

LIST OF TABLES

Table	Title	Page
1	Summary of Graphical Data	53
2	Characteristic Equations (for Computer Programming)	76

LIST OF ILLUSTRATIONS

Figure		Page
1.	Effect of Pair of Complex Poles near the Imaginary Axis on High Gain Systems (Bode Plot)	8
2.	Root Loci from Complex Pole with Adjacent Complex Zero	9
3.	Angles of Emergence Defined	12
4.	Effect of Zero Angular Position on Angle of Emergence	16
5.	Effect of Zero Radial Displacement on Angle of Emergence	18
6.	Relationship between Residue Angle of Complex Pole and Complex Zero	21
7.	Angle of Emergence vs. Radial Position	23
8.	Angle of Emergence vs. Zero Position Angle, θ	24
9.	Case Study Pole-Zero Configurations	26
10.	Root Locus Data	31
11.	Root Locus Data (Computer)	73
12.	Root Locus Data (Computer)	74
13.	Root Locus Data (Computer)	75

TABLE OF SYMBOLS AND ABBREVIATIONS

- α - angle defined in Fig. 6 and equations (2-16) and (2-17)
- β - constant angle defined by equation (2-7) and (2-8)
- η - angle defined in Fig. 6 and equations (2-16) and (2-17)
- \ominus - angular position of complex zero with respect to its neighboring complex pole
- $\angle k_p$ - residue angle of complex pole of "uncompensated" system (no complex zeros)
- $\angle k_{pc}$ - residue angle of complex pole with adjacent "compensating" zeros
- $\angle k_z$ - residue angle of complex zero
- λ - angle defined in Fig. 4
- ϕ - angle defined in Fig. 5
- ψ_c - angle of emergence from a complex pole with adjacent "compensating" complex zero
- ψ_p - angle of emergence from a complex pole ("uncompensated system")
- ψ_z - angle of emergence from a complex zero

CHAPTER I

INTRODUCTION

Many closed loop transfer functions of control systems such as those representing structural resonances in missiles, fuel sloshing, shaft twisting, etc., have a pair of complex conjugate poles near the imaginary axis. If gain requirements are not too high, the effect of the roots near these poles is often neglected because other roots dominate. However, in the design of high gain systems, these roots near the imaginary axis may cross over into the right half s -plane and thus cause instability in the system. Figure 1 shows the effect of raising gain on the open-loop Bode diagram.

Cancel compensation would be an effective solution to removing these complex poles near the $j\omega$ -axis if it were not for (1) the difficulty in the design accuracy of these compensators, and (2) the poles themselves may not be fixed, and may move about in an area in the s -plane.

The mere placing of complex zeros near the complex poles is no assurance that the root loci is connected between these poles and zeros. (See Fig. 2) When this loci does close on the adjacent complex zero, the root on this locus has such a small residue that it can be neglected when compared with the other roots.

The question remains: What are the requirements for the location of complex zeros adjacent to complex poles for best cancel compensation? The search for an answer to this question initiated this study. Such

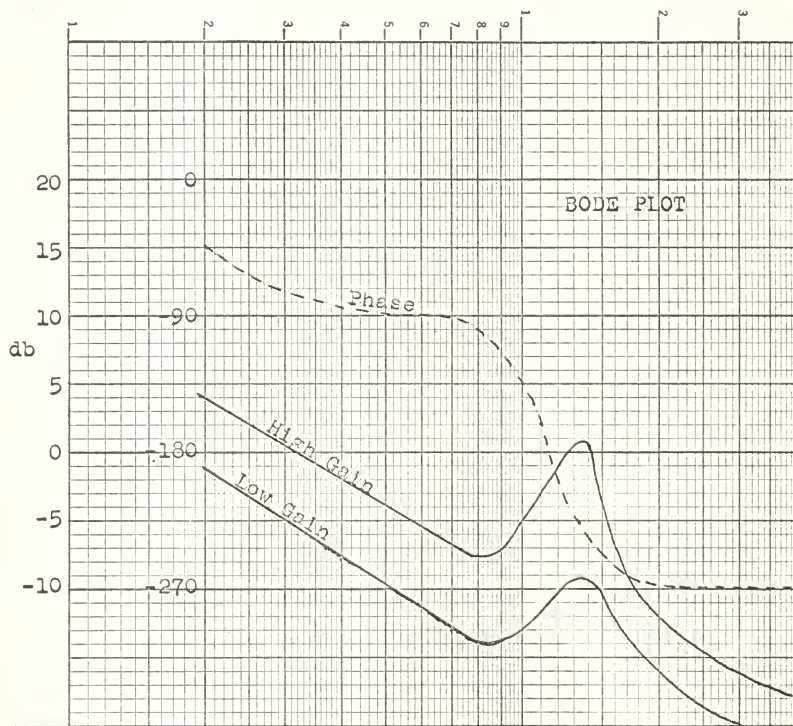


Fig. 1

Effect of a pair of Complex Poles near the Imaginary Axis on High Gain Systems

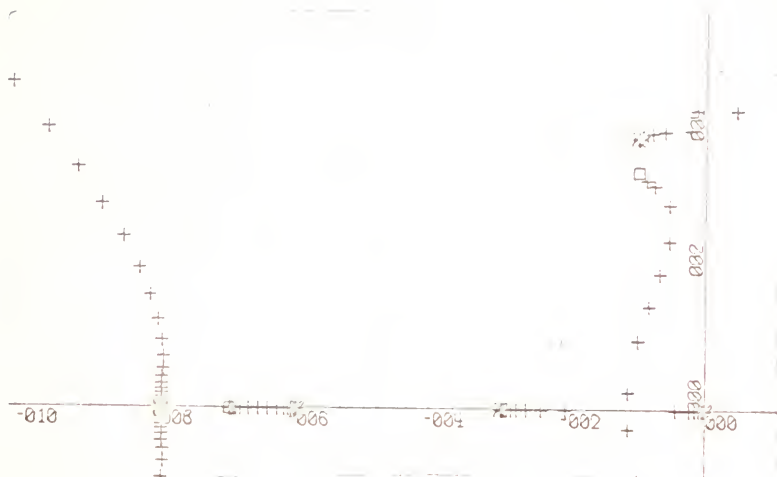


Fig. 2

Root Loci from Complex Pole with adjacent Complex Zero

a location or area might make it possible to locate complex zeros for the design of self-adaptive compensators.

First in this study, the angle of emergence from a complex pole and zero was studied to determine how it varied and how it was related to the transfer function.

Next, 43 different case studies were programmed into the Control Data Corporation 1604 Computer to obtain 215 root locus plots. These plots and observations are contained in Chapter 3.

The term "compensated system" will be defined hereafter to mean a system with a pair of complex conjugate zeros relatively close to complex conjugate poles. The "uncompensated system" will be defined as the system without the complex conjugate zeros, but will include the real poles associated with a "zero-compensator."

CHAPTER II

ANGLE OF EMERGENCE

1. Determination of the Angle of Emergence

The angle of emergence of the root locus from a complex pole or zero has been defined in the literature as the angle of the tangent to the root-locus curve at the pole or zero as shown in Fig. 3. A test point s_1 near the complex pole on the root locus satisfies the equation

$$\sum_{i=1}^N \angle Z_i - \sum_{i=1}^M \angle P_i - \angle P_1 = 180 + n(180)$$

where n is an odd integer or zero, N is the total number of zeros, M is the total number of poles, and $\angle Z_1$ is the angle of the complex pole infinitesimally close to the test point. Fig. 3 shows that $\angle P_1$ is measured counter-clockwise with the negative abscissa axis as reference, whereas the angle of emergence, ψ_p is measured counterclockwise with the positive abscissa axis as reference. It is thus seen that the angle of emergence is equal to the angle from the test point to the complex pole, using their corresponding references. Thus,

$$\psi_p = \angle P_1 = 180 + \sum_{i=1}^N \angle Z_i - \sum_{i=1}^M \angle P_i \quad (2-1)$$

The last two terms in the above equation can be determined with the complex pole or zero (from which the angle of emergence is to be determined) as the origin of the coordinate system, since the test point s_1 is infinitesimally close to the complex pole (or zero). Thus, the origin of the s -plane is translated to the desired complex pole or zero by setting $w = s + \alpha - j\beta$ where w is the new complex variable in

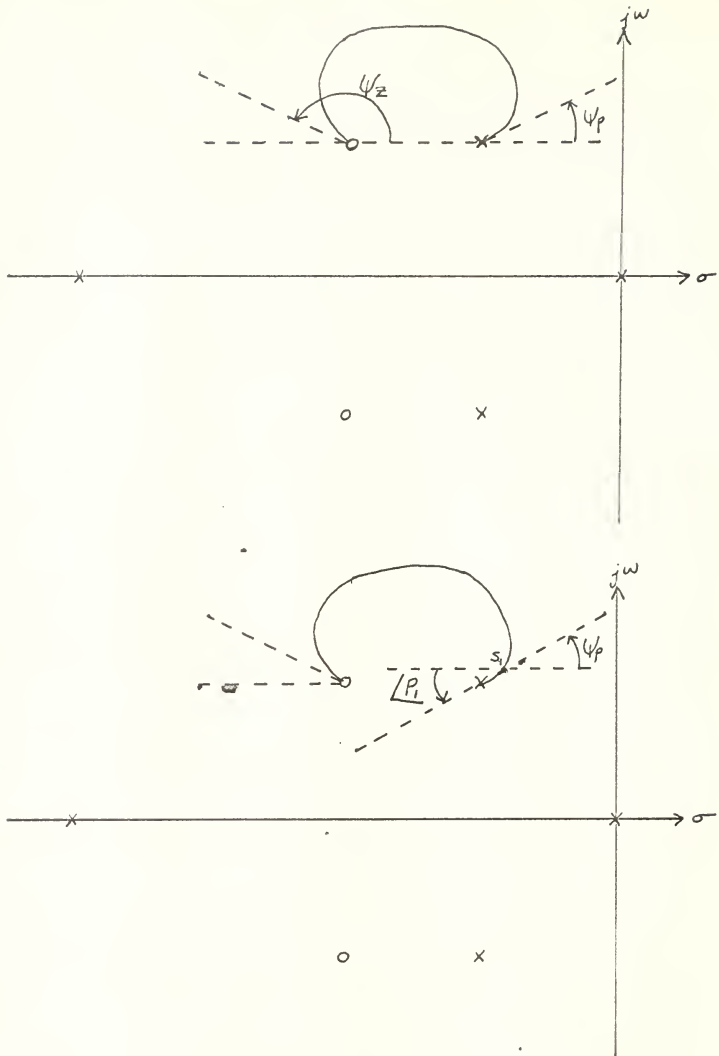


Fig. 3

Angle of Emergence of complex pole, ψ_p
 Angle of Emergence of complex zero, ψ_z
 Angle from test point s_1 to complex pole, $\angle P_1$

the w-plane. A test point w_1 near the new origin (pole or zero) which is on the root-locus must still satisfy the angle criterion (180 degrees or an odd multiple). As $|w|$ approaches zero the sum of the angles on the root locus plus 180 degrees would be the angle from the origin to the test point. When $|w|$ is identically zero, there is no angular contribution from the pole or zero at the origin and thus the sum of the remaining angles is the angle of emergence. For example, a characteristic equation is partitioned as follows:

$$\frac{(s+2-j5)(s+3+j5)}{s(s+10)(s+2+j5)(s+2-j5)} = -1$$

To find the angle of emergence at the complex pole located at $-2+j5$ in the s-plane, transfer the origin to this pole by setting $w = s+2-j5$ or $s = w-2+j5$. Then,

$$\frac{(w-2+j5+3-j5)(w-2+j5+3+j5)}{(w-2+j5)(w-2+j5+10)(w-2+j5-2-j5)(w-2+j5-2+j5)} = -1$$

$$\frac{(w+1)(w+1+j10)}{(w-2+j5)(w+8+j5)(w)(w+j10)} = -1$$

$$w = \frac{(w+1)(w+1+j10)}{(-1)(w-2+j5)(w+j10)(w+8+j5)} \quad (2-2)$$

Define the angle of emergence from the complex pole,

$$\psi_p = \angle_{|w|=0} w \quad (2-2a)$$

$$\psi_p = \tan^{-1} \frac{0}{1} + \tan^{-1} \frac{10}{1} - (180 + \tan^{-1} \frac{5}{-2} + \tan^{-1} \frac{5}{8} + 90)$$

$$\psi_p = 0 + 84.5 - (180 + 111.6 + 32 + 90)$$

$$\psi_p = 30.9$$

Now to find the angle of emergence from the complex zero located at $-3+j5$ in the s-plane, transfer the origin to this zero by setting $w = s+3-j5$ or $s = w-3+j5$

$$\frac{(w-3+j5+3-j5)(w-3+j5+3+j5)}{(w-3+j5)(w-3+j5+10)(w-3+j5+2-j5)(w-3+j5+2+j5)} = -1$$

$$\frac{(w)(w+j10)}{(w-3+j5)(w+7+j5)(w-1)(w-1+j10)} = -1$$

$$w = \frac{(-1)(w-3+j5)(w+7+j5)(w-1)(w-1+j10)}{(w+j10)} \quad (2-3)$$

Define the angle of emergence from the complex zero,

$$\psi_z = \angle w \quad |w| = 0 \quad (2-3a)$$

$$\psi_z = 180 + \tan^{-1} \frac{5}{-3} + \tan^{-1} \frac{5}{7} + \tan^{-1} \frac{0}{-1} + \tan^{-1} \frac{10}{-1} - 90$$

$$\psi_z = 180 + 121 + 35.5 + 180 + 95 - 90$$

$$\psi_z = 161.5$$

2. Complex Residues and the Angle of Emergence

The closed loop transfer function can be expanded by partial fraction expansion and the residues evaluated by use of Heaviside's expansion theorem: (a) multiply function by $(s - p_k)$, and (b) let $s = p_k$, giving the residue at K_{pk} . Using the previous example again, the residue angle at

$$s = -2 + j5 \text{ is: } \angle K_{rc} = \frac{(s+3-j5)(s+3+j5)}{s(s+10)(s+2+j5)} \bigg|_{s = -2+j5} \quad (2-4)$$

It is seen that this residue evaluation is similar to equation (2-1) except for the -1 in the denominator which introduces 180 degrees. Thus the angle of the residue at a complex pole differs from the angle of emergence by 180 degrees:

$$\psi_{pc} = \angle K_{rc} + 180 \quad (2-5)$$

The characteristic equation of the preceding section may be partitioned as follows, where the zeros are now the "poles" and the poles are now the "zeros":

$$\frac{s(s+10)(s+2-j5)(s+2+j5)}{(s+3-j5)(s+3+j5)} = -1$$

The left side of this equation is seen to be the reciprocal of the closed-loop transfer function. Although this is an improper fraction, the residues can still be evaluated by Heaviside's expansion theorem. Thus, the residue at $s = -3+j5$ is:

$$\angle K_z = \frac{s(s+10)(s+2-j5)(s+2+j5)}{(s+3+j5)} \quad s = -3+j5 \quad (2-6)$$

$$\angle K_z = \frac{(-3+j5)(-3+j5+10)(-3+j5-2-j5)(-3+j5+2+j5)}{(-3+j5+3+j5)} \quad (2-7)$$

$$\angle K_z = 341.5$$

Comparing equations (2-7) and (2-3) it is seen that they differ by 180 degrees. Thus the angle of emergence from a complex zero is

$$\psi_z = \angle K_z + 180 \quad (2-7a)$$

where $\angle K_z$ is determined as illustrated above.

3. Effect of Complex Zeros near Complex Poles

What effect does the relative position of complex zeros adjacent to complex poles have on the angle of emergence from the complex poles? To answer this question, first consider the change of the angle of emergence with the zeros at constant radius from the poles while varying the angular position of the zeros with respect to the complex poles.

It is assumed that the displacement between complex poles and the distance between complex pole and its neighboring zero is of a ratio of approximately 10:1. As seen in Fig. 4, as the angle θ , the angular position of the

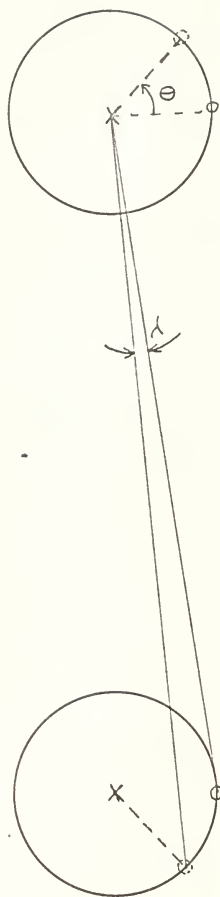


Fig. 4

Effect of Zero Angular Position on Angle of Emergence

complex zero, increases, the angle of emergence increases in magnitude. The angular contribution of λ due to the lower half plane zero is negligible compared with θ in the upper half plane. Therefore a linear slope of one exists between angular position of zero and angle of emergence, or

$$\frac{\partial \psi}{\partial \theta} \approx 1 \quad (2-8)$$

Next consider Fig. 5 which shows the effect of radial position of complex zeros (with constant angular position) on the angle of emergence. As the zeros vary in radial position, there is no angular change due to upper half plane zero. The change in lower half plane zero angle is ϕ . In addition to the assumption stated above, the line P_1Z_2 is assumed perpendicular to line P_2A for small changes in radial position r . Then

$$\phi = \tan^{-1} \left(\frac{r \cos \theta}{d - r \sin \theta} \right) \quad \text{where } d \text{ is the displacement distance between complex poles, } P_1P_2$$

$$\frac{\partial \phi}{\partial r} = \frac{\frac{\partial}{\partial r} \left(\frac{r \cos \theta}{d - r \sin \theta} \right)}{1 + \frac{r^2 \cos^2 \theta}{(d - r \sin \theta)^2}}$$

$$\frac{\partial \phi}{\partial r} = \frac{[(r \cos \theta)(-1)(d - r \sin \theta)^{-2}(-\sin \theta) + (d - r \sin \theta)^{-1} \cos \theta]}{1 + \frac{r^2 \cos^2 \theta}{(d - r \sin \theta)^2}}$$

$$\frac{\partial \phi}{\partial r} = \frac{\left[\frac{r \sin \theta \cos \theta}{(d - r \sin \theta)^2} + \frac{\cos \theta}{d - r \sin \theta} \right]}{1 + \frac{r^2 \cos^2 \theta}{(d - r \sin \theta)^2}}$$

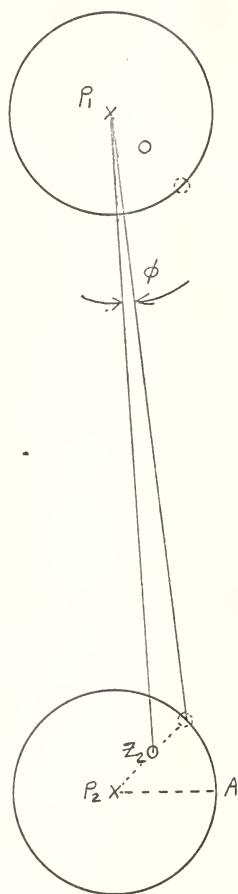


Fig. 5

Effect of Zero Radial Displacement on Angle of Emergence

$$\frac{\partial \phi}{\partial r} = \frac{r \sin \theta \cos \theta + \cos \theta (d - r \sin \theta)}{(d - r \sin \theta)^2 + r^2 \cos^2 \theta}$$

$$\frac{\partial \phi}{\partial r} = \frac{d \cos \theta}{d^2 - 2dr \sin \theta + r^2}$$

Since it was assumed that $d/r = 10$, $d^2 \gg -2dr \sin \theta + r^2$

$$\frac{\partial \phi}{\partial r} = \frac{\cos \theta}{d} = \quad (2-9)$$

The total derivative of the angle of emergence is:

$$d\psi = \frac{\partial \psi}{\partial \theta} d\theta + \frac{\partial \psi}{\partial r} dr$$

For small displacements,

$$\psi_c = \theta + \frac{r \cos \theta}{d} + \beta \quad (2-10)$$

where ψ_c is the angle of emergence of the compensated system, and β is a constant. If r , the distance between adjacent complex poles and zeros, is allowed to become zero, or "perfect cancellation" with poles, the angular position of the zeros is undefined. The constant β is then evaluated as the sum of all remaining zeros in the system minus the sum of all remaining poles, measured from the upper half plane "cancelled" complex pole. Thus in the example cited in section 1, β is evaluated as 210.9. For $\theta = 180$, the angle of emergence $\psi_p = 30.9$ degrees. The second term on the right side of equation (2-10) is negligible compared with the other terms under the assumptions that have been made. (See Fig. 7) It is seen that the angle β differs from the residue angle of the uncompensated system (no complex zeros present) by 90 degrees.

$$\angle K_p = \beta - 90 \quad (2-11)$$

The angle of emergence of the uncompensated system, ψ_p , is readily determined:

$$\psi_p = K_p + 180 \quad (2-5a)$$

$$\psi_p = \beta - 90 + 180 = \beta + 90 \quad (2-12)$$

Equation (2-7) can now be written as:

$$\psi_z = \theta + \frac{\cos \theta}{d} r + \psi_p - 90$$

$$\psi_z \cong \theta + \psi_p - 90 \quad (2-13)$$

Thus the angle of emergence changes directly with the zero position angle θ . (See Fig. 8)

The residue angles of the complex pole and zero are defined in equations (2-4) and (2-7) respectively and can be written in the following form::

$$\angle K_{pc} = \frac{\angle z_1 \angle z_2}{\angle p_1 \angle p_2 \angle p_3} \quad \left| \text{at pole } p_4 \right. \quad (2-14)$$

$$\angle K_z = \frac{\angle p_1 \angle p_2 \angle p_3 \angle p_4}{\angle z_1} \quad \left| \text{at zero } z_2 \right. \quad (2-15)$$

The relationship between the pole residue angle, K_{pc} , and the zero residue angle K_z can now be written as:

$$\angle K_z = -\angle K_{pc} + \angle z_2 + \angle p_4 \quad (2-16)$$

Let $\alpha = \angle z_2$ and $\eta = \angle p_4$ The equation (2-16) becomes:

$$\angle K_z = -\angle K_{pc} + \alpha + \eta \quad (2-17)$$

The angles α and η are seen in Fig.6. It is observed from this figure that $\alpha = \theta - 180$ and $\eta = \theta$

Substituting these relationships into equation (2-17) yields the following result:

$$\angle K_z = -\angle K_{pc} + 2\theta - 180 \quad (2-18)$$

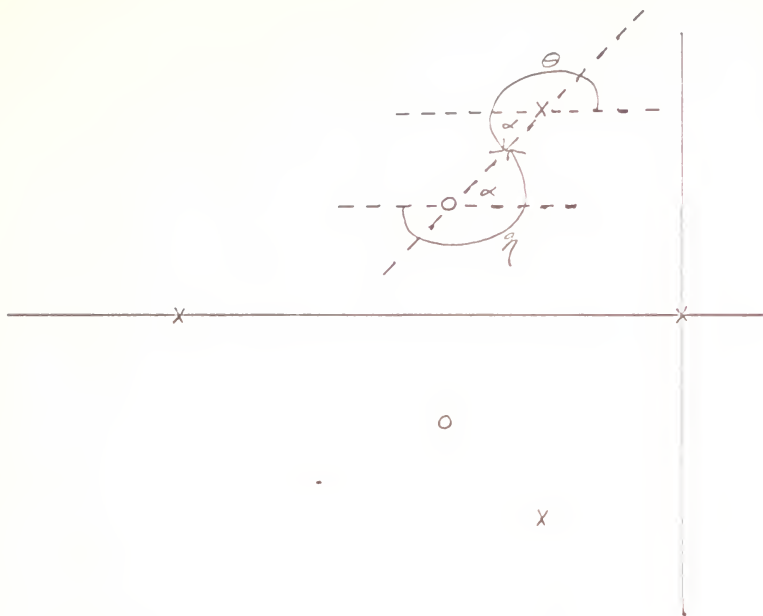


Fig. 6

Relationship between Residue Angle of Complex Pole and Complex Zero.

$$\angle K_z = \angle K_p + \alpha + \gamma$$

But from equations (2-5) and (2-7a),

$$\angle K_{pc} = \psi_c - 180$$

$$\angle K_z = \psi_z - 180$$

Therefore,

$$\begin{aligned}\psi_z - 180 &= -\psi_c + 180 + 2\theta - 180 \quad \text{or,} \\ \psi_z &= -\psi_c + 2\theta + 180\end{aligned}\tag{2-19}$$

Substituting equation (2-13) into equation (2-19) yields the relation between the angle of emergence from the complex zero and the angle of emergence from the complex pole when the complex zeros are not present.

$$\begin{aligned}\psi_z &= -\theta - \psi_p + 90 + 2\theta + 180 \\ \psi_z &= \theta - \psi_p - 90\end{aligned}\tag{2-20}$$

Using the relation from equation (2-5a), then

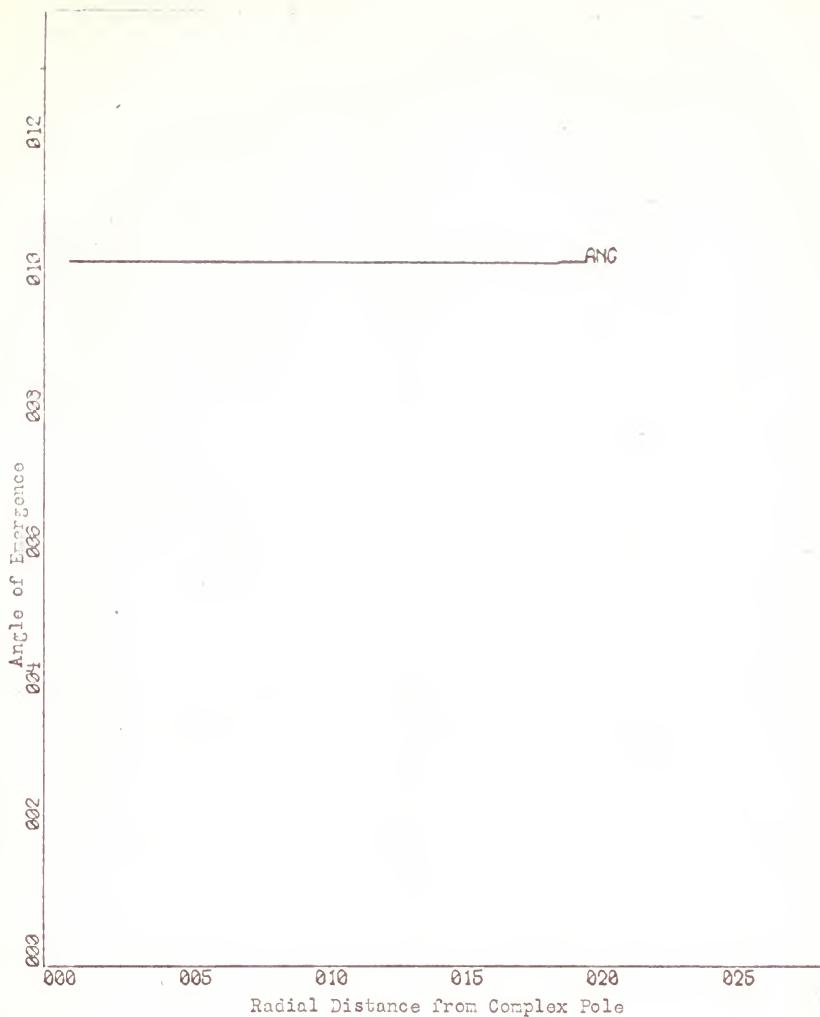
$$\psi_z = \theta - \angle K_p + 90\tag{2-21}$$

Subtracting equation (2-13) from equation (2-20) gives the relation:

$$\begin{aligned}\psi_z - \psi_c &= -2\psi_p = -2(\angle K_p + 180) \\ \psi_z - \psi_c &= -2\angle K_p\end{aligned}\tag{2-22}$$

Thus the difference between the angle of emergence from a complex pole and the angle of emergence of its neighboring complex zero is a constant for a given system, and thus not a function of zero position θ .

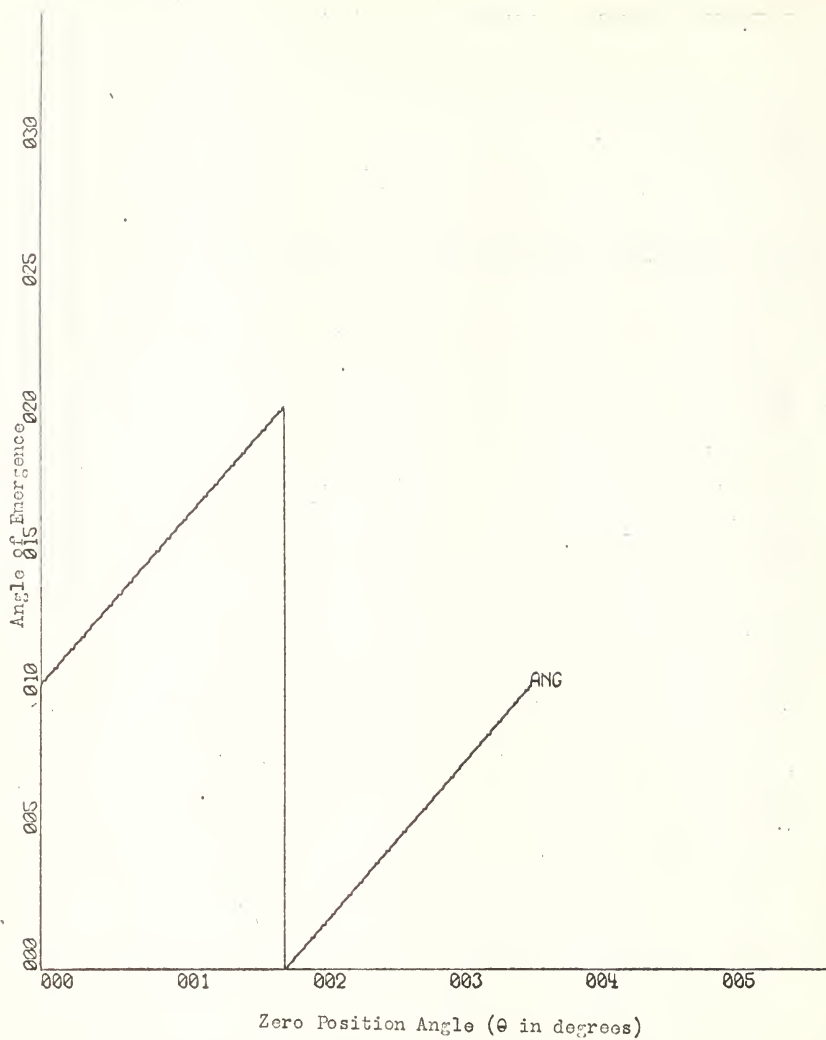
Figures 7 and 8 are computer-solution graphs of the angle of emergence versus complex zero radial and angular positions respectively.



X-SCALE - 5.00E-01 UNITS/INCH
 Y-SCALE - 2.00E+01 UNITS/INCH

Fig. 7

Angle of Emergence vs. Radial Position of Complex Zero (θ constant)



X-SCALE - 1.00E+02

Y-SCALE - 5.00E+01

Fig. 8

Angle of Emergence vs. Zero Position Angle, θ

CHAPTER III

ROOT LOCI LEAVING COMPLEX POLES

1. Data Designation

Having analyzed the angles of emergence from complex poles and zeros, the next step in this study is to determine the effect of various system transfer functions on the root loci emerging from the complex poles, and to determine when this loci closes on its adjacent complex zero, i.e., the root loci starts at the complex pole and ends on the complex zero. As shown in Fig.2 of Chapter I, this loci may go to the zero at infinity instead of the complex zero.

Over fifty different cases were tried and the data consisted of about 300 root locus plots using the computer program in Appendix I. Only 43 case studies (Fig. 9) are presented with 215 individual root locus plots (Fig. 10) for this analysis.

Each case study or individual root locus plot is designated or classified using letters and numbers, such as "A-1a", where the first (capital) letter designates generally the number of real poles and zeros, and position of complex conjugate poles. The number indicates the variable position of the given number of real poles and zeros. The last small letter specifies the variable position of the complex zero with respect to the complex pole.

Double, triple, etc., real poles are specified by placing an "X" over an "X" or a group of "X's" near appropriate point on the plot.

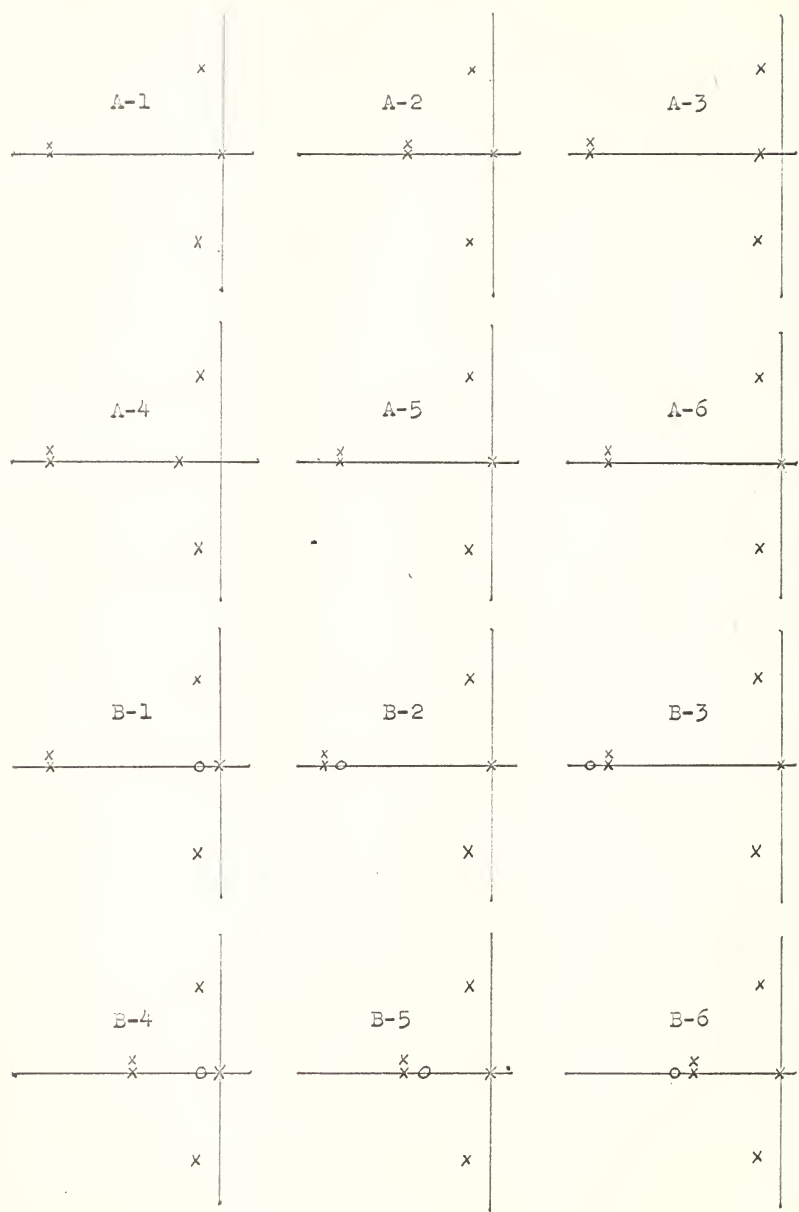


Fig. 9a Case studies

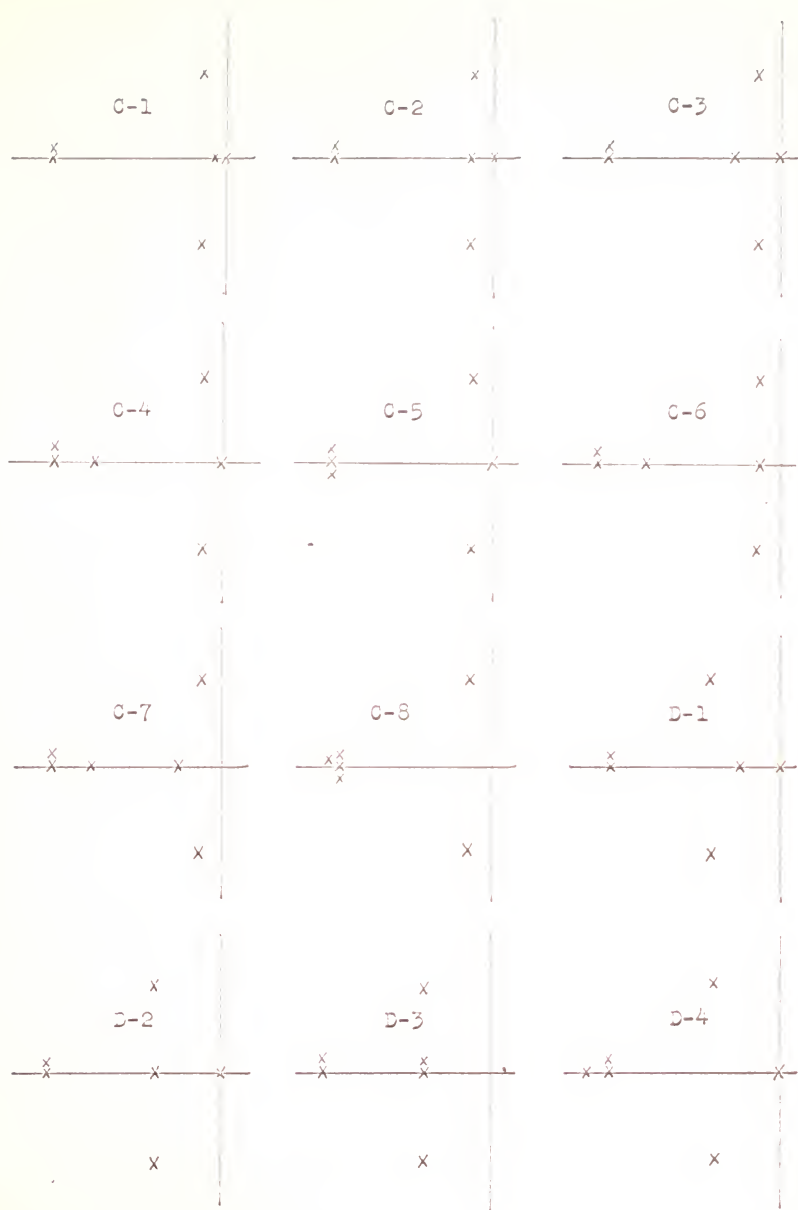


Fig. 9b Case studies

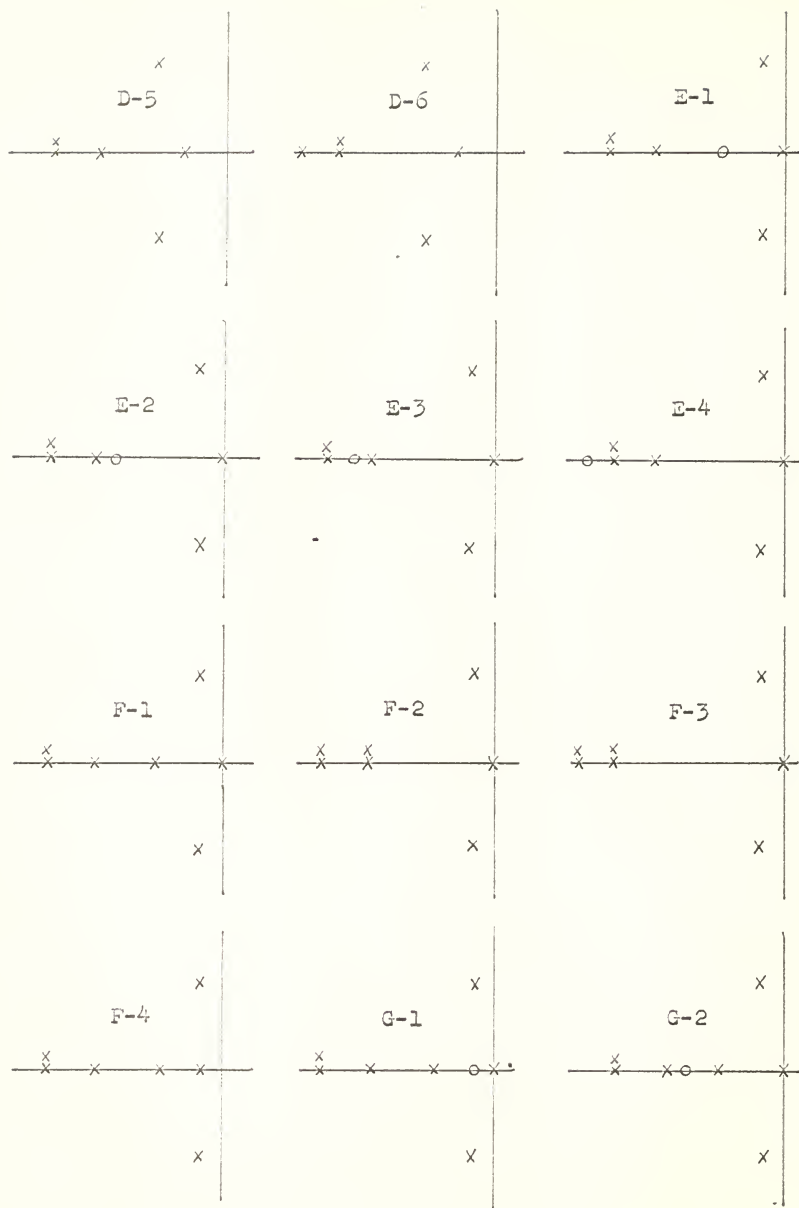


Fig. 9c Case studies

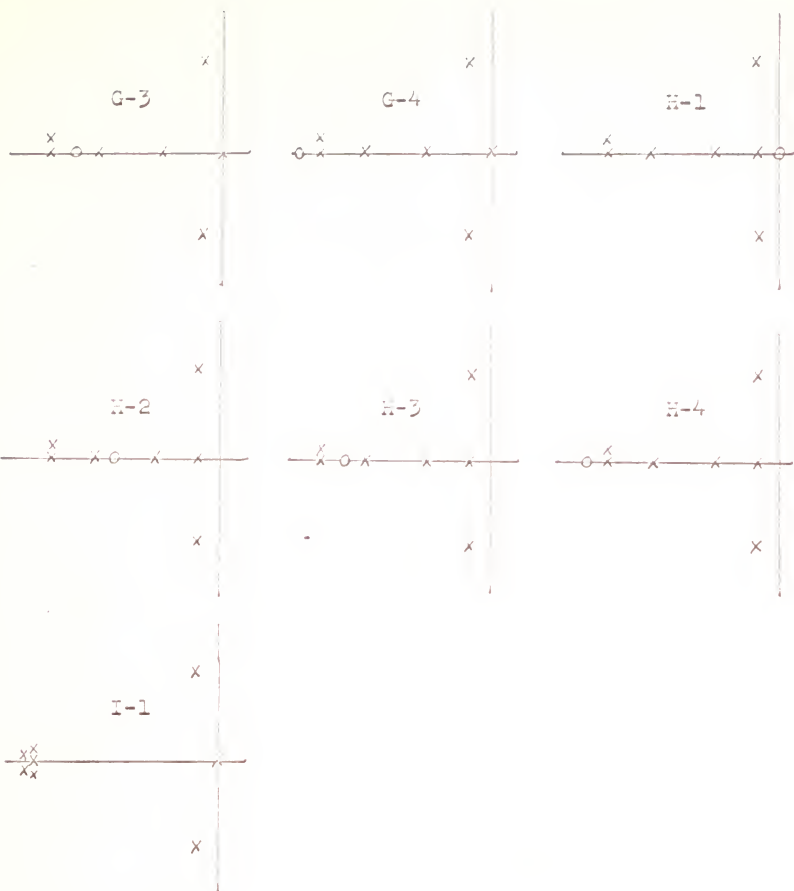


Fig. 9d Case studies

The double real poles are arbitrarily used to represent the poles associated with a "cancel-compensator" and is usually located, for this study, at -8 units except in a few cases they are placed at -4 units.

Case A is a fifth order system. Case B is a fifth order system with one real zero. Case C is a sixth order system. Case D is a sixth order system with the complex conjugate poles moved out to $-3 \pm j4$. Case E is a sixth order system with one real zero. Case F is a seventh order system. Case G is a seventh order system with one real zero. Case H is also a seventh order system with one real zero. Case I is an eighth order system. Figure 9 presents these 43 case studies, and shows only the pole-zero configurations without the complex conjugate zeros. Figure 10 shows the root locus plots for these case studies when the complex zeros are placed in various positions.

Table 1 is a summary of the graphical data from Fig. 10. In addition it contains the computed values of angles of emergence from pole and zero (for $\theta = 0^\circ$) and the uncompensated residue angle using the formulas developed in Chapter II. A comparison with these calculated values from the transfer function compares very good with computer root locus plots.

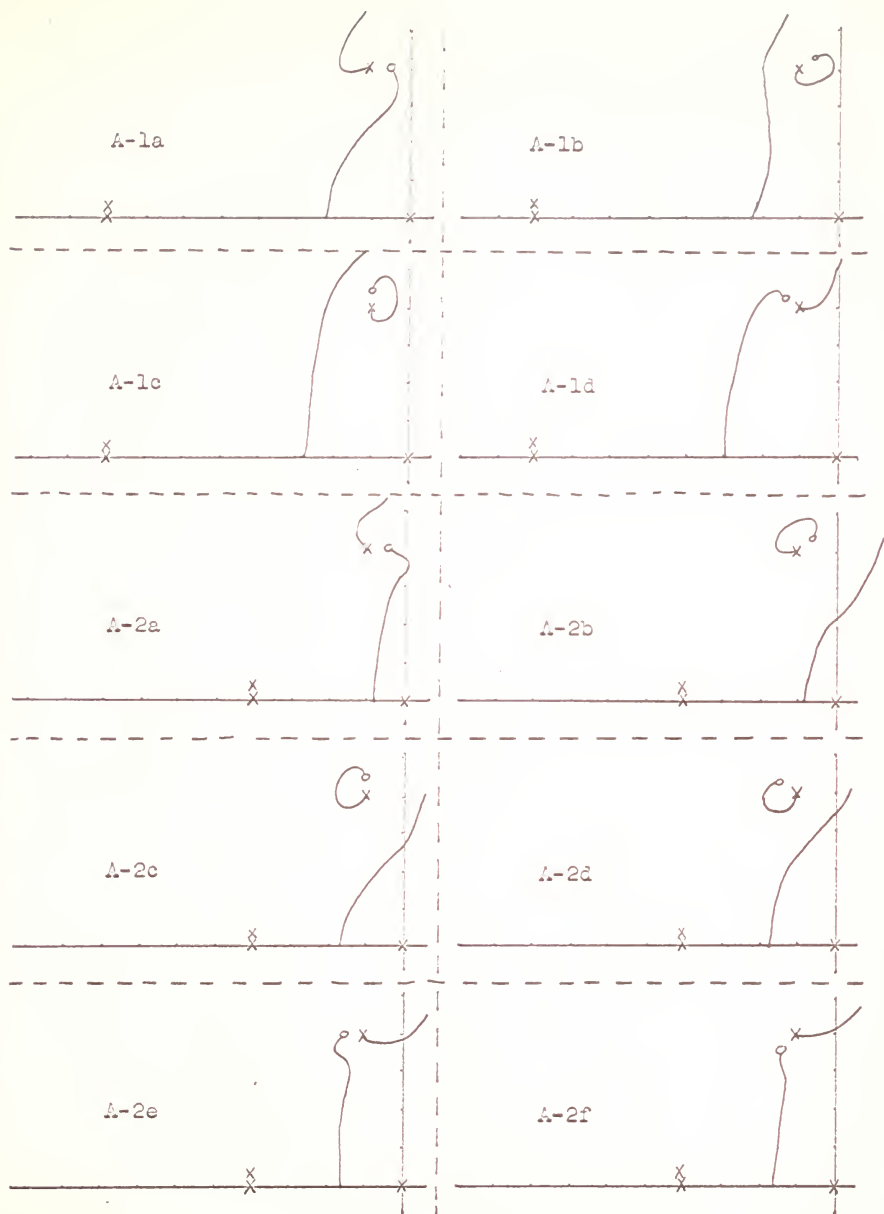


Fig. 10a Root Locus Data

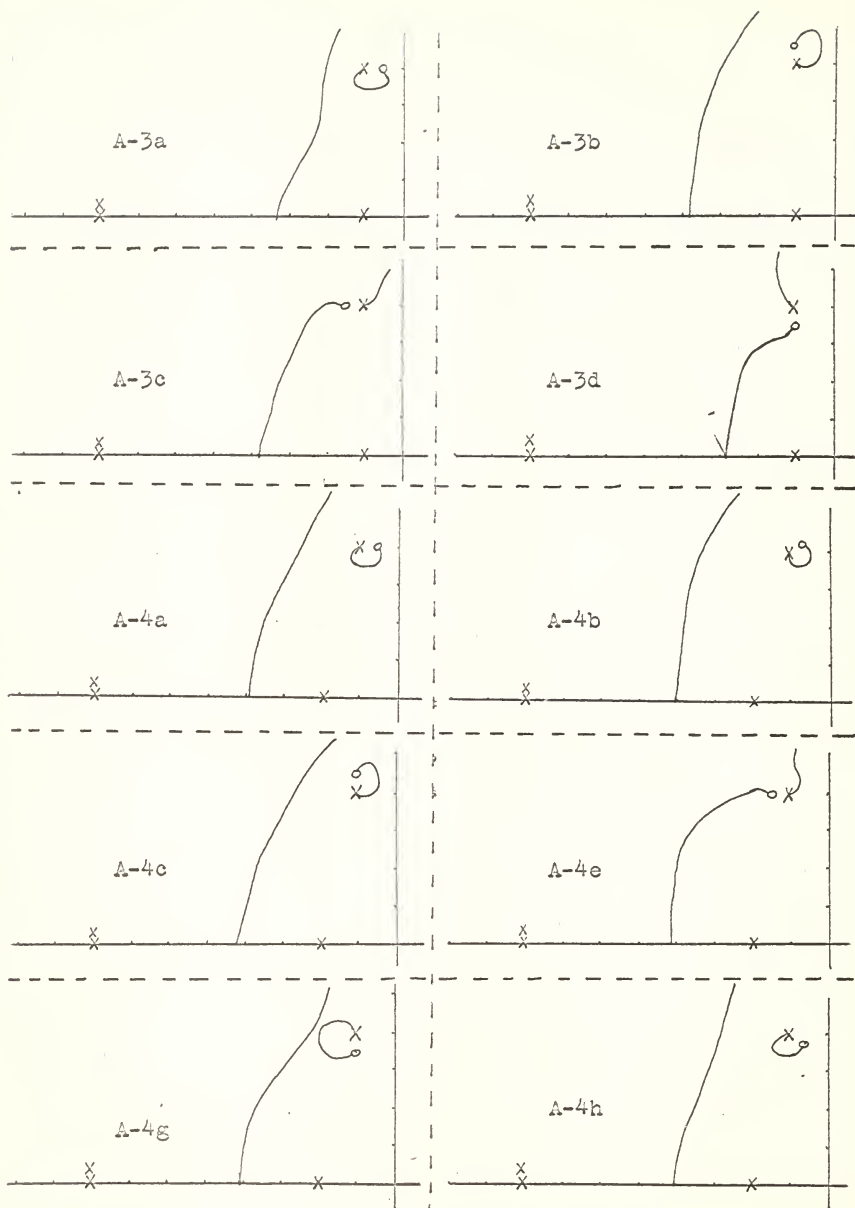


Fig. 10b Root Locus Data

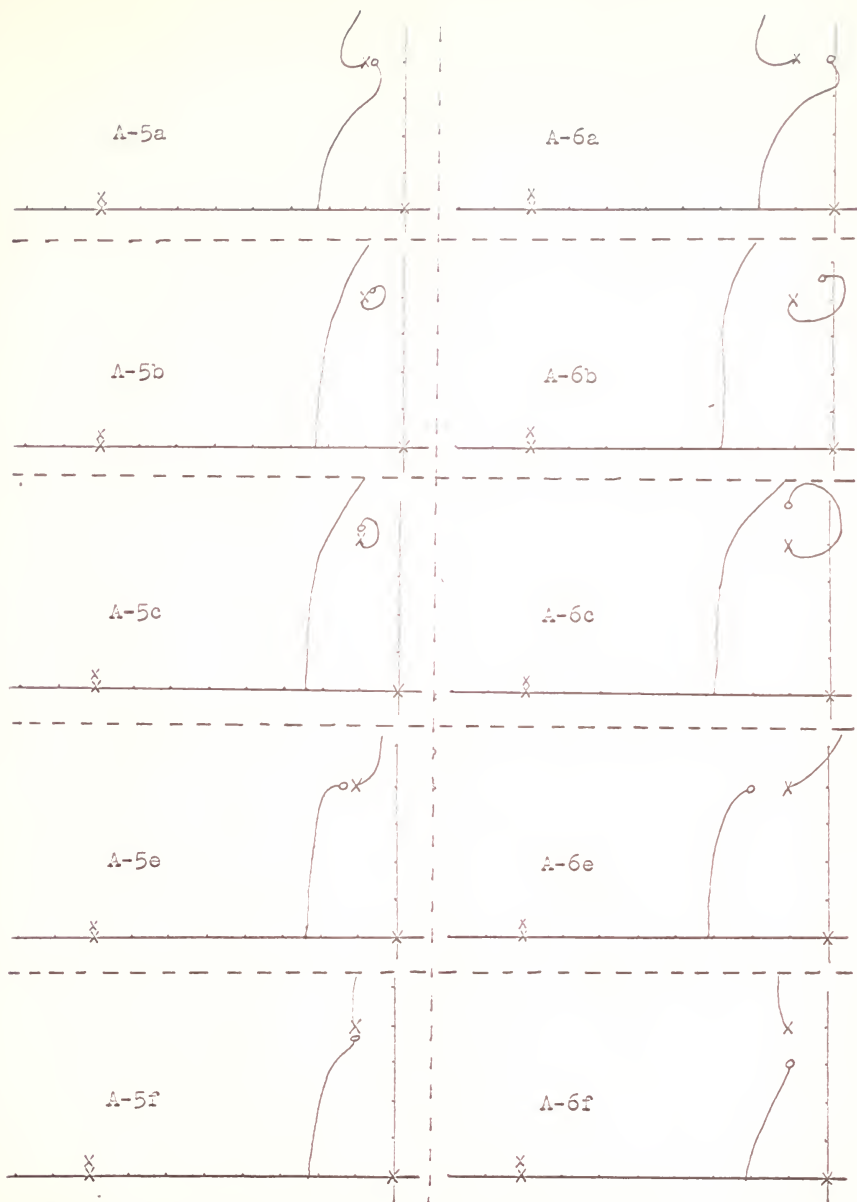


Fig. 10c Root Locus Data

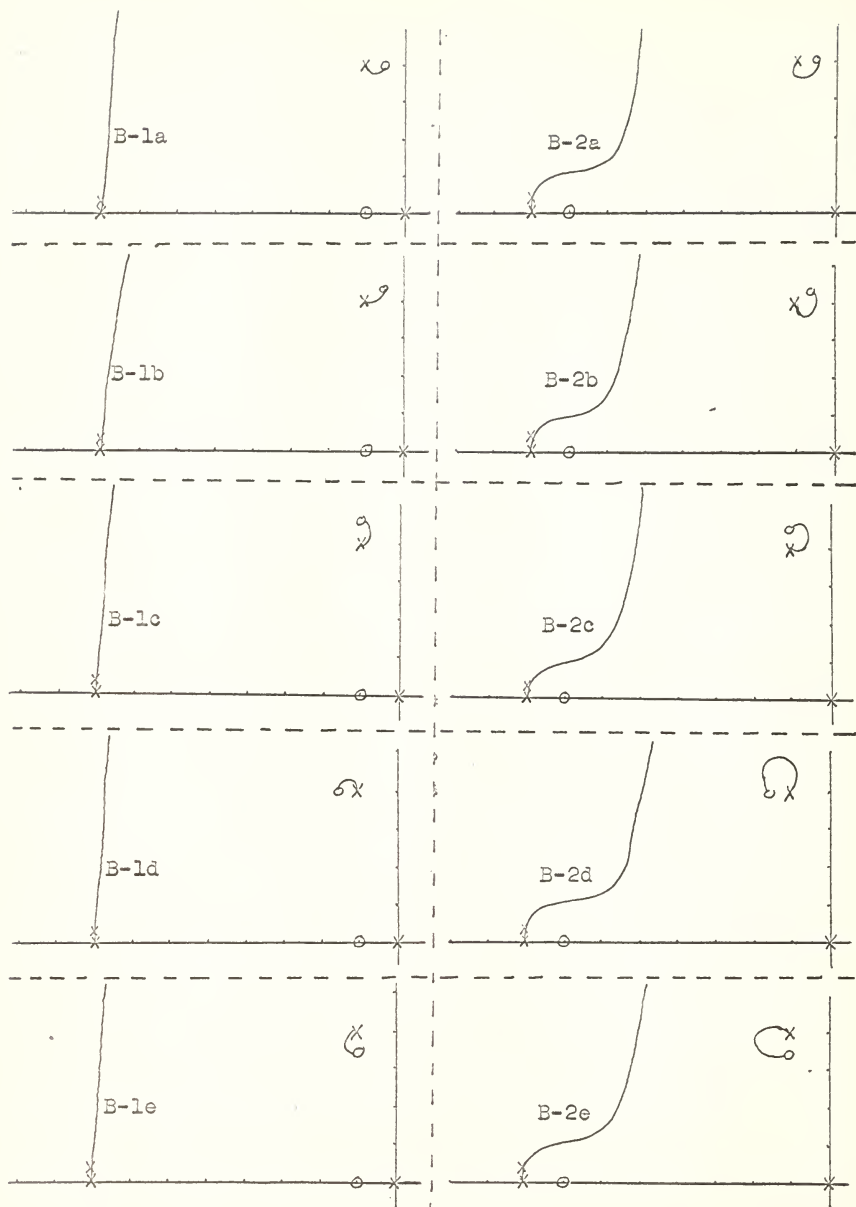


Fig. 10d Root Locus Data

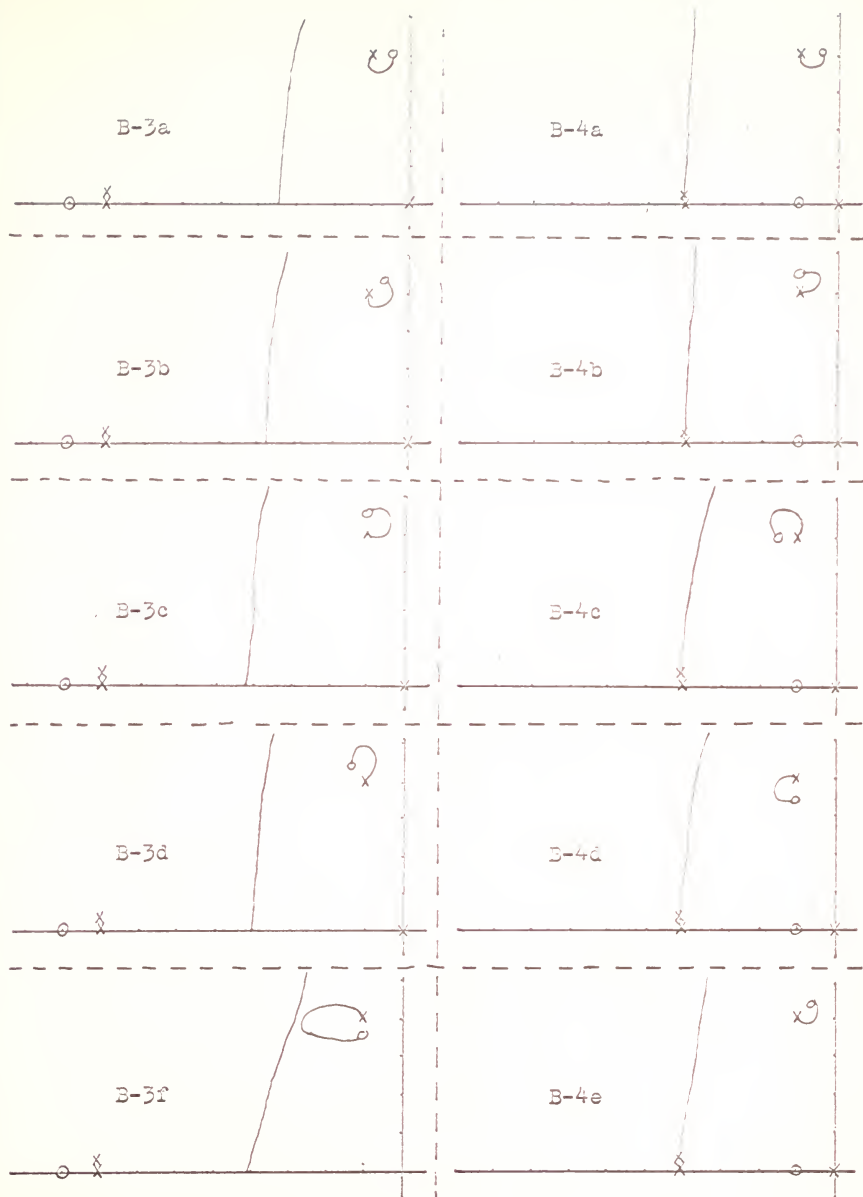


Fig. 10e Root Locus Data

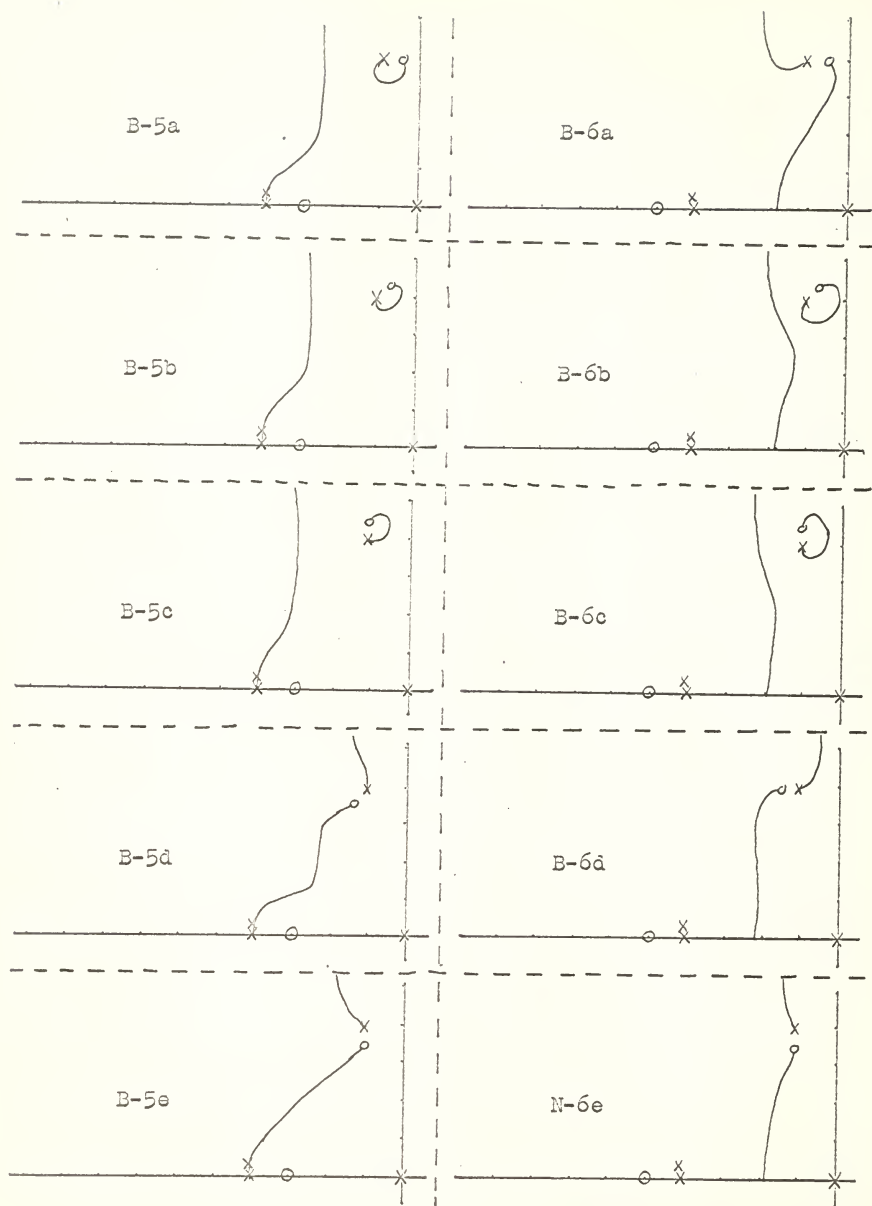


Fig. 10f Root Locus Data

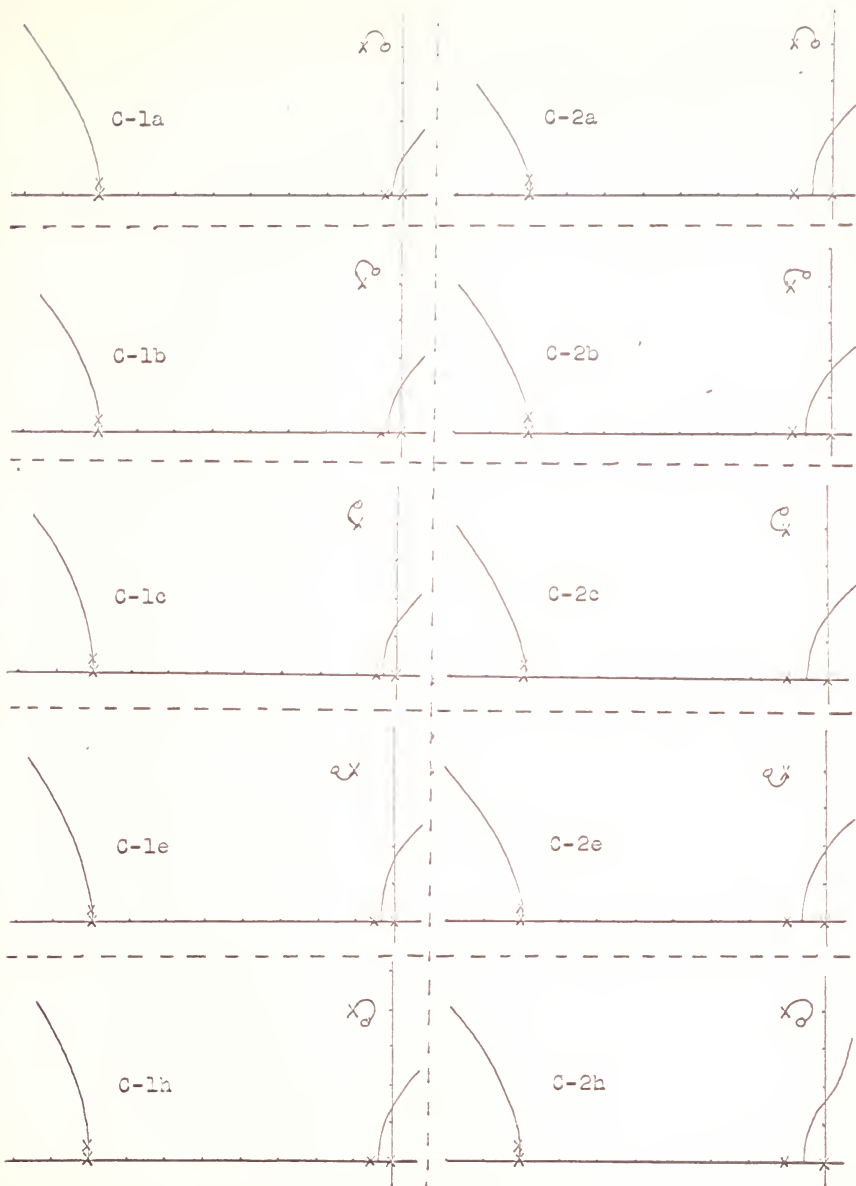


Fig. 10g Root Locus Data

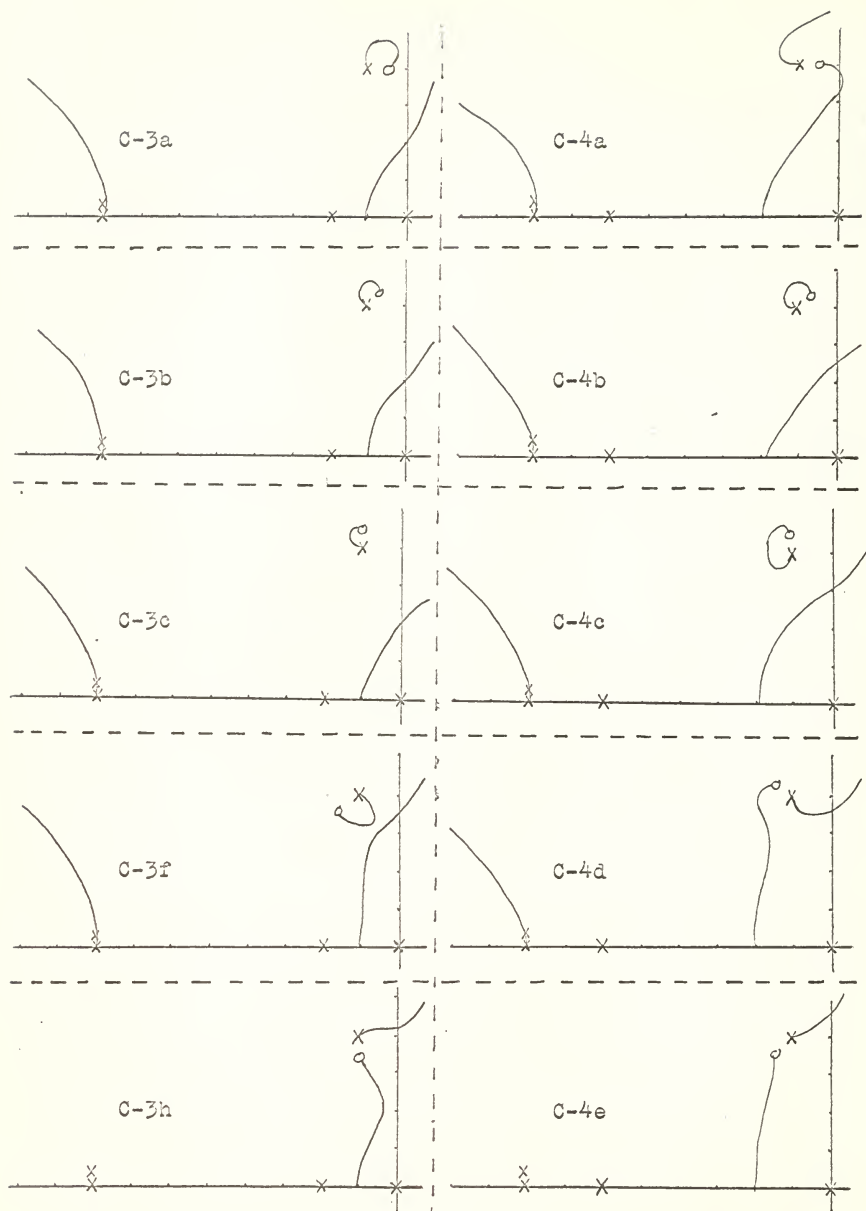


Fig. 10h Root Locus Data

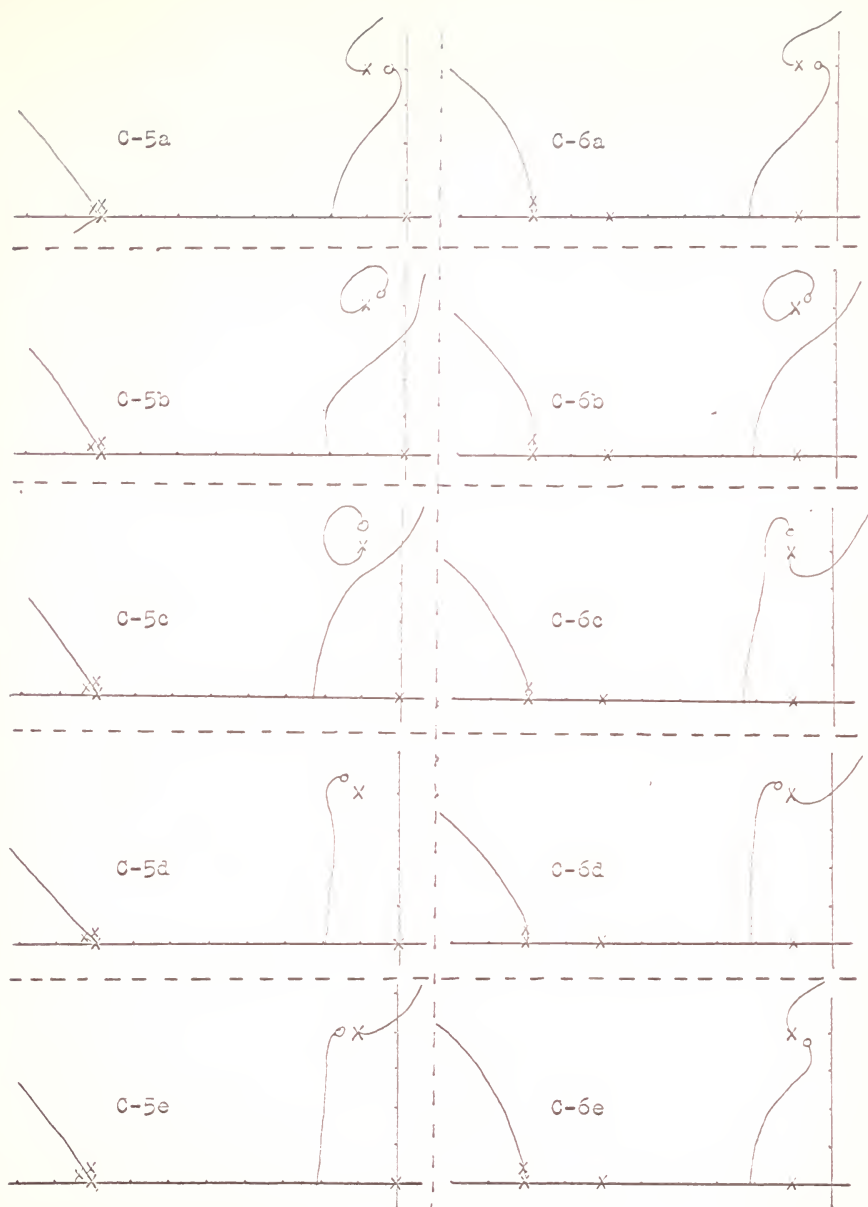


Fig. 10i Root Locus Data

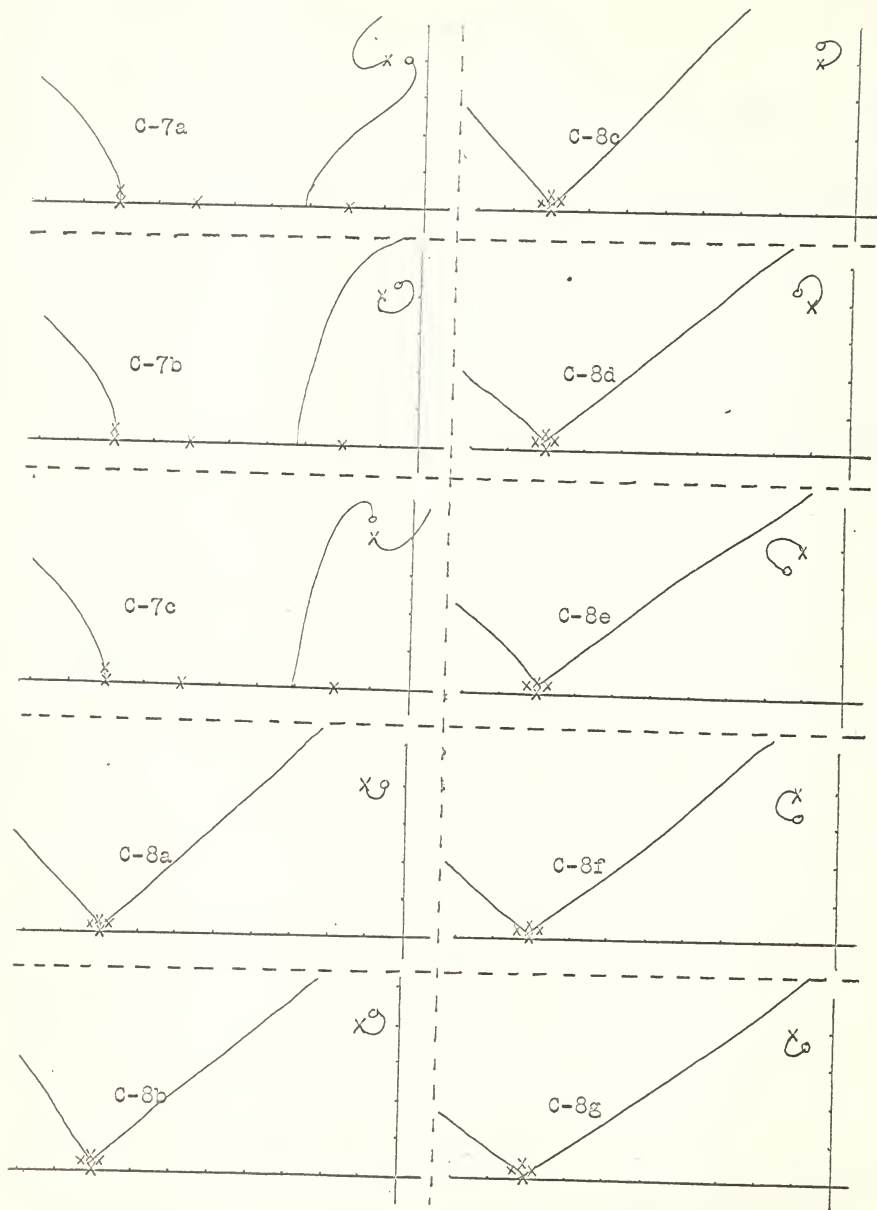


Fig. 10j Root Locus Data

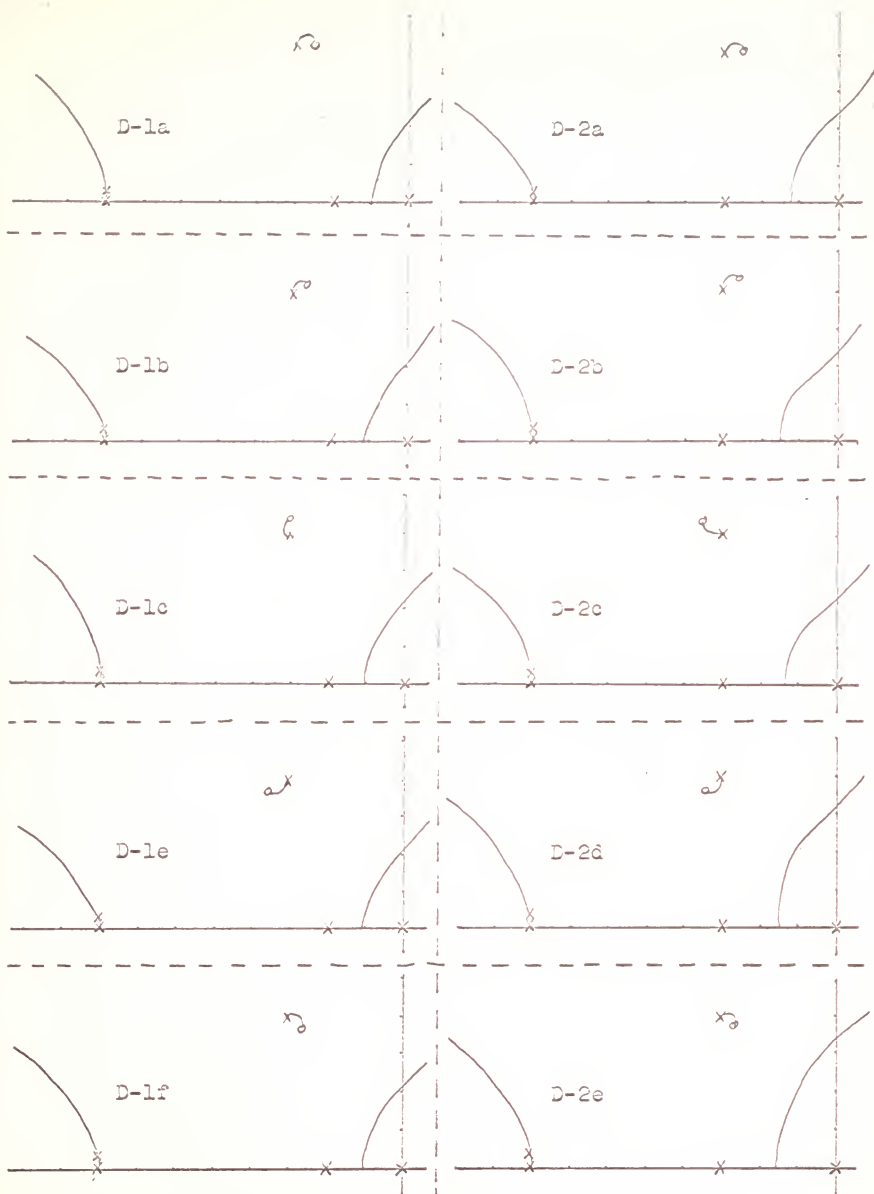


Fig. 10k Root Locus Data

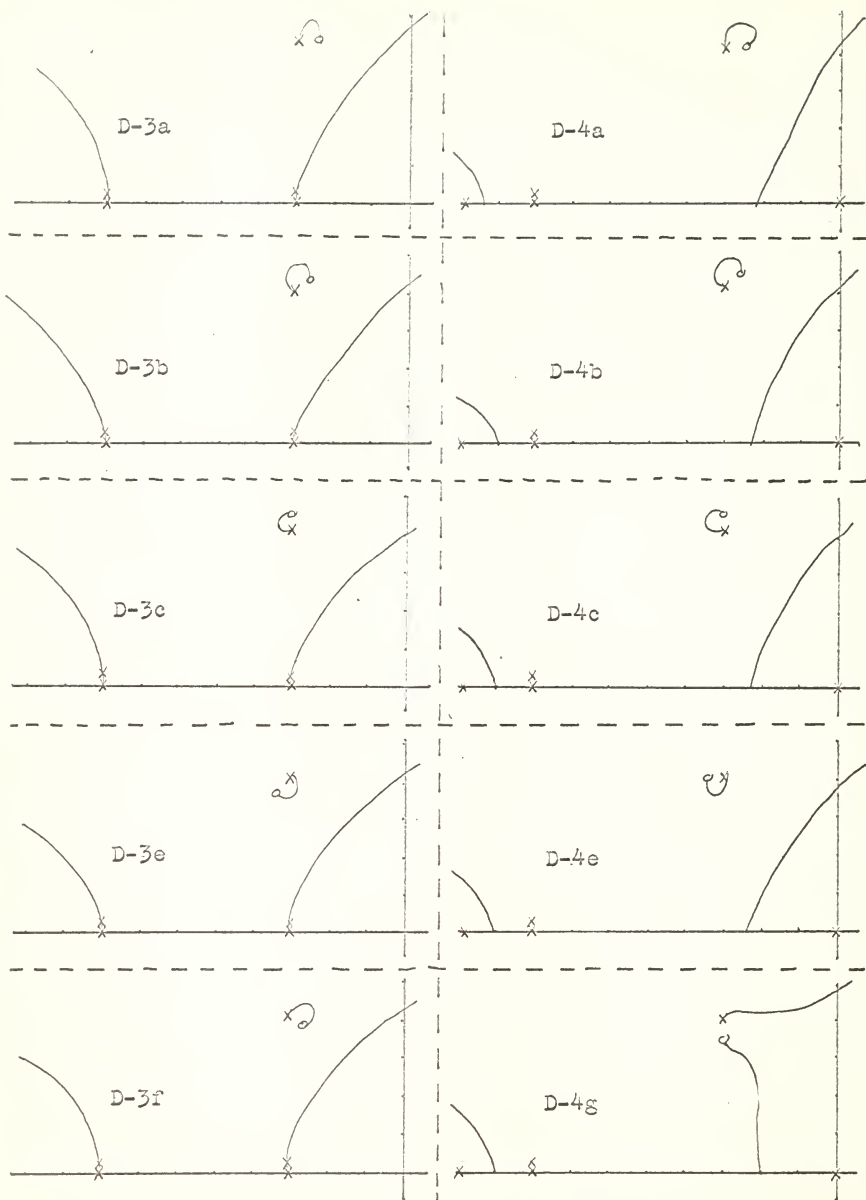


Fig. 101 Root Locus Data

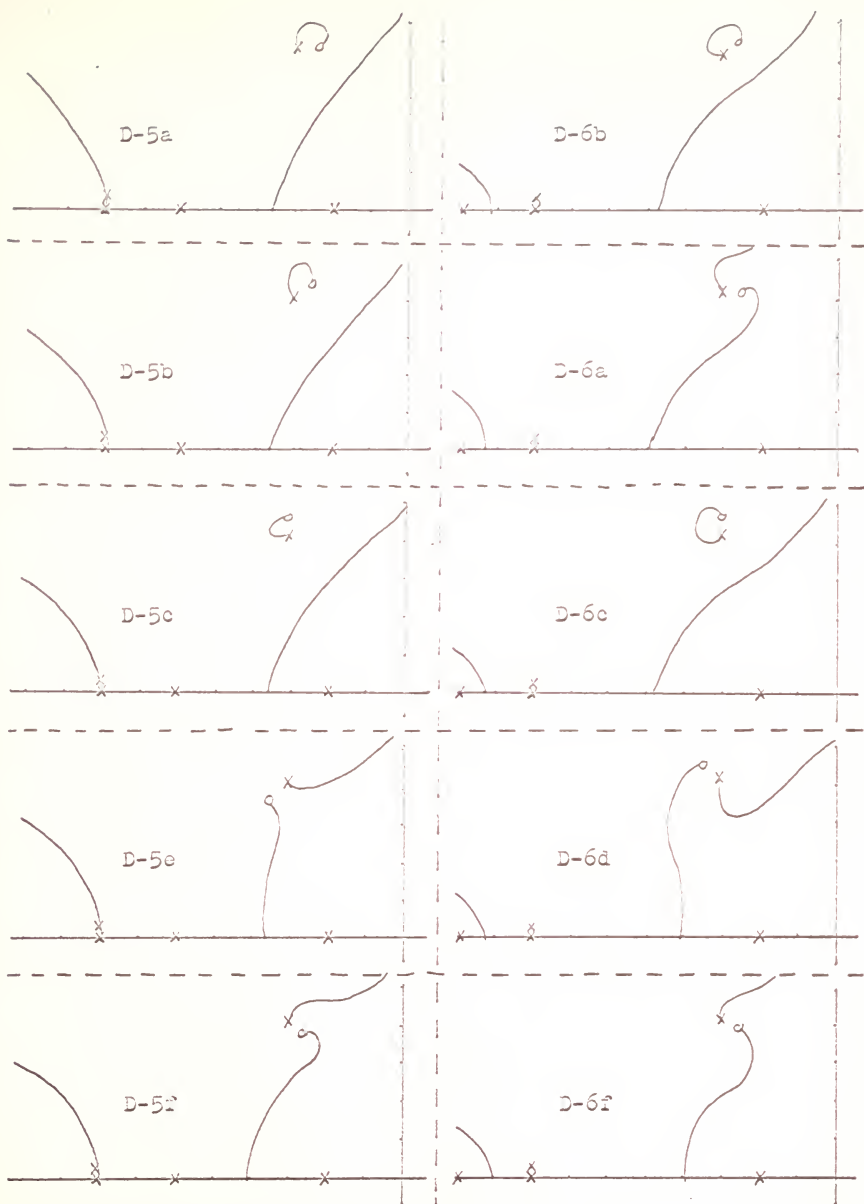


Fig. 10m Root Locus Data

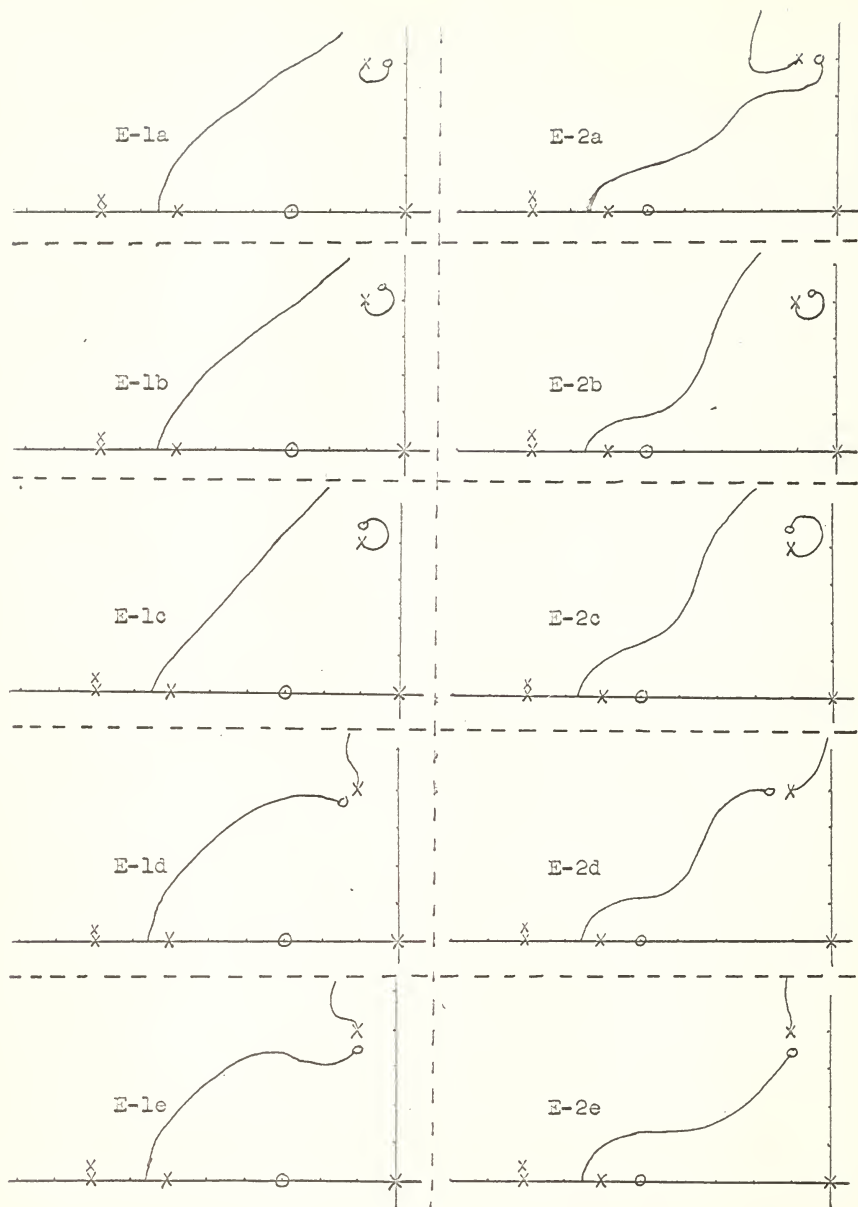


Fig. 10n Root Locus Data

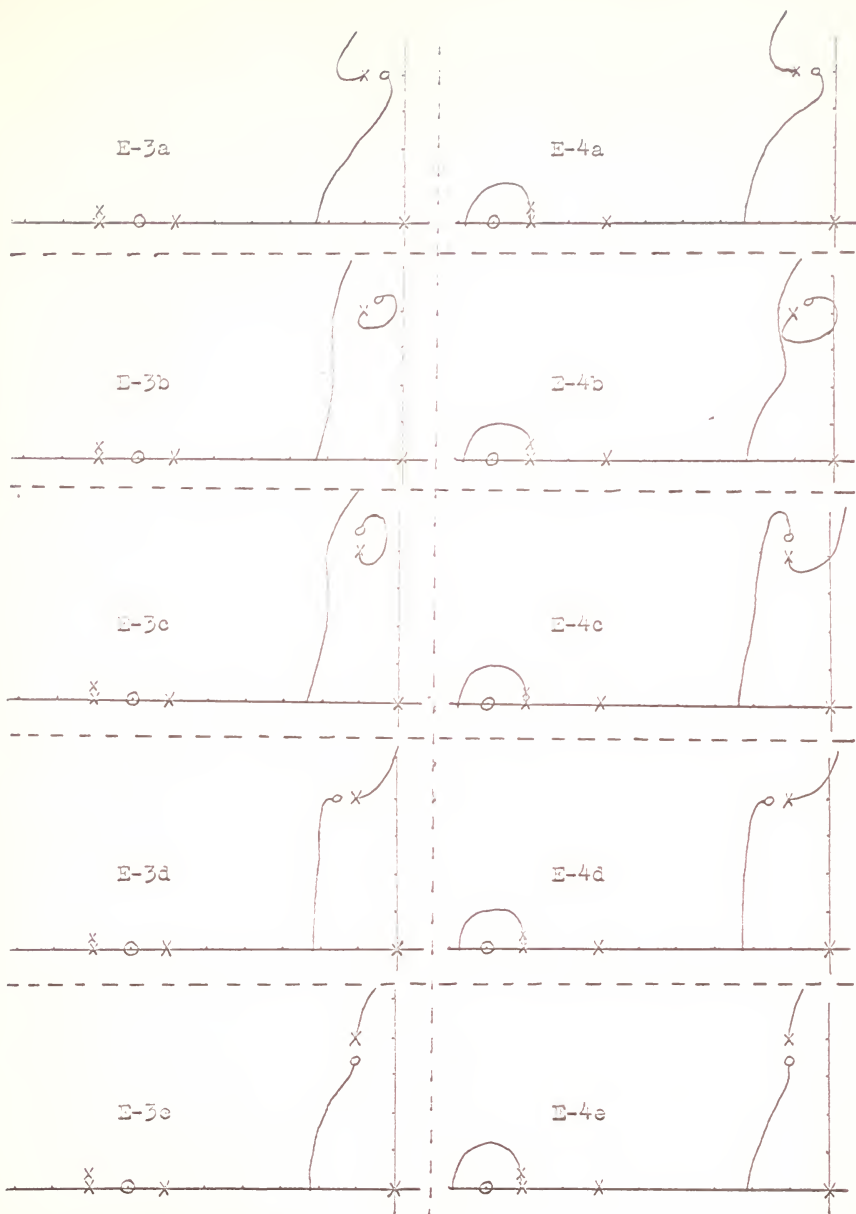


Fig. 10o Root Locus Data

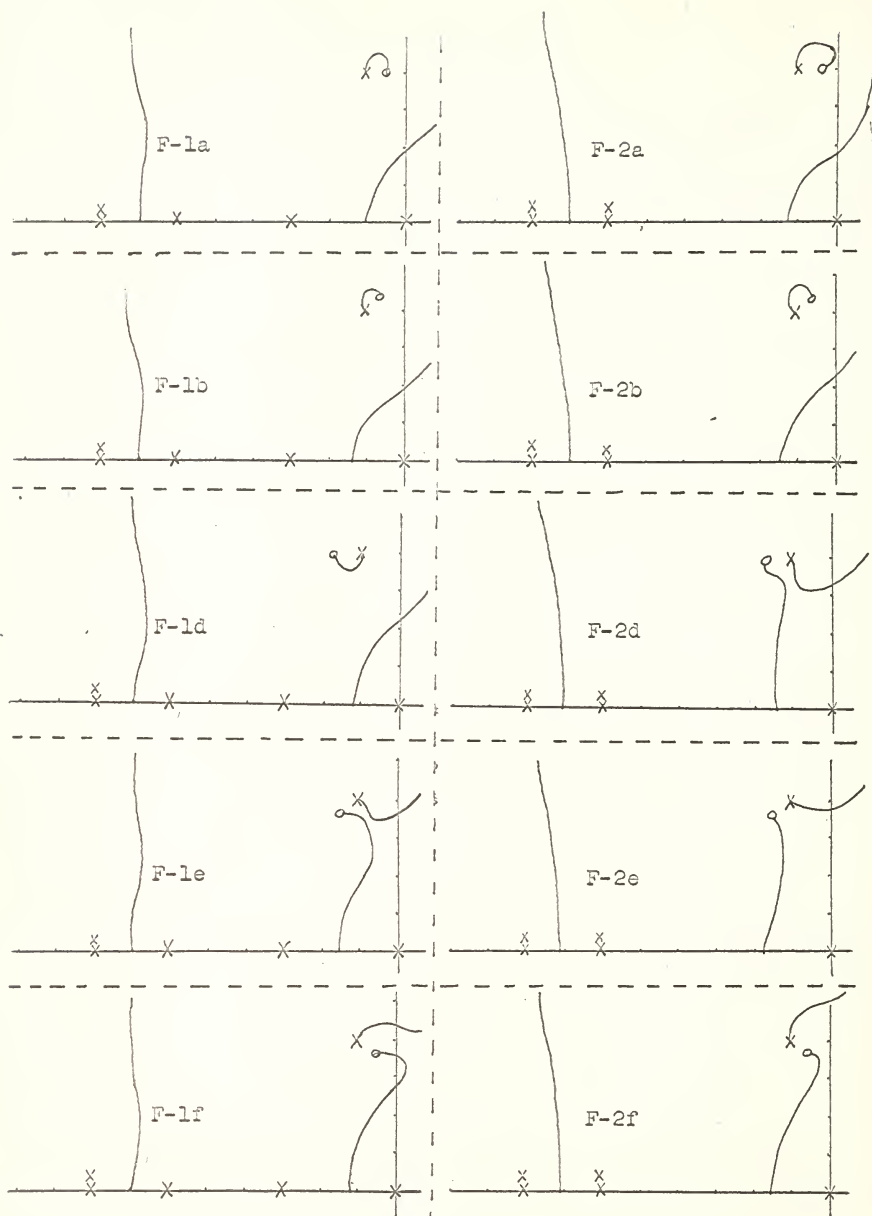


Fig. 10p Root Locus Data

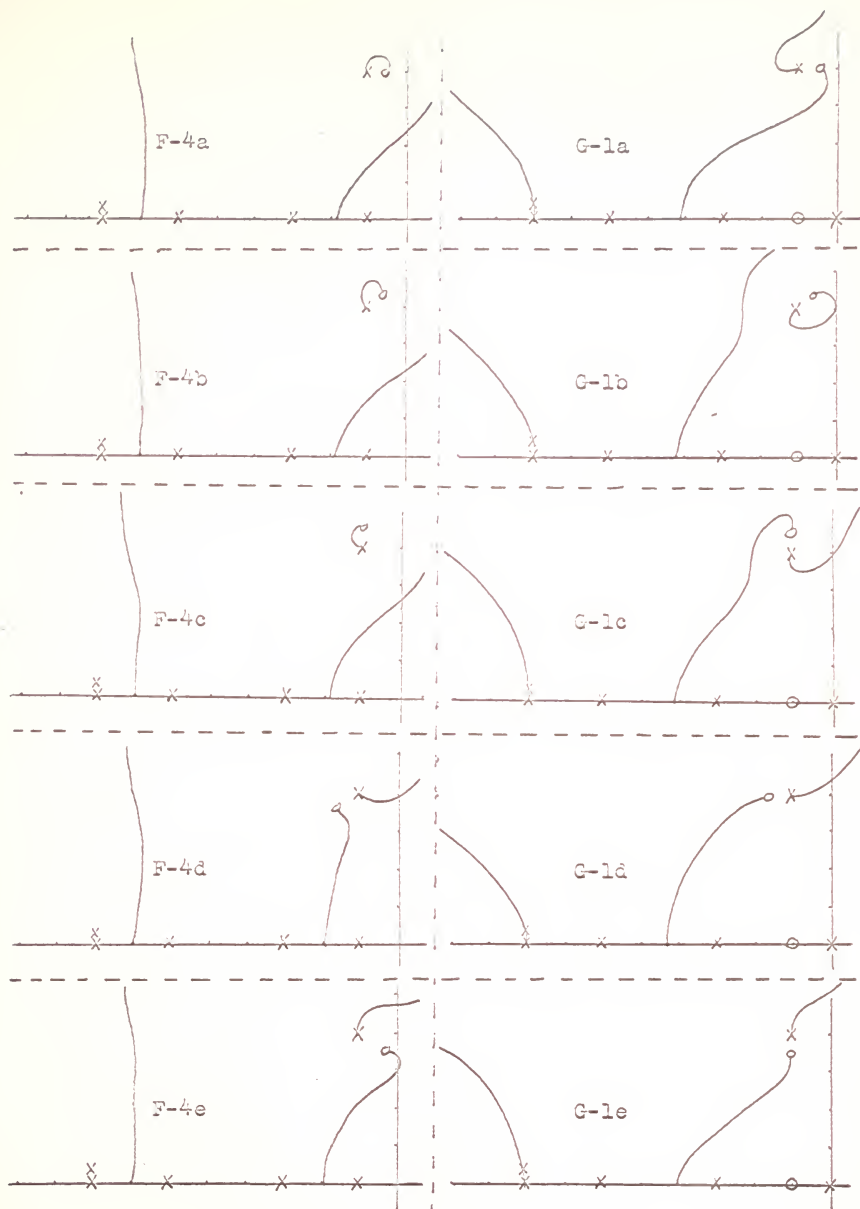


Fig. 10q Root Locus Data

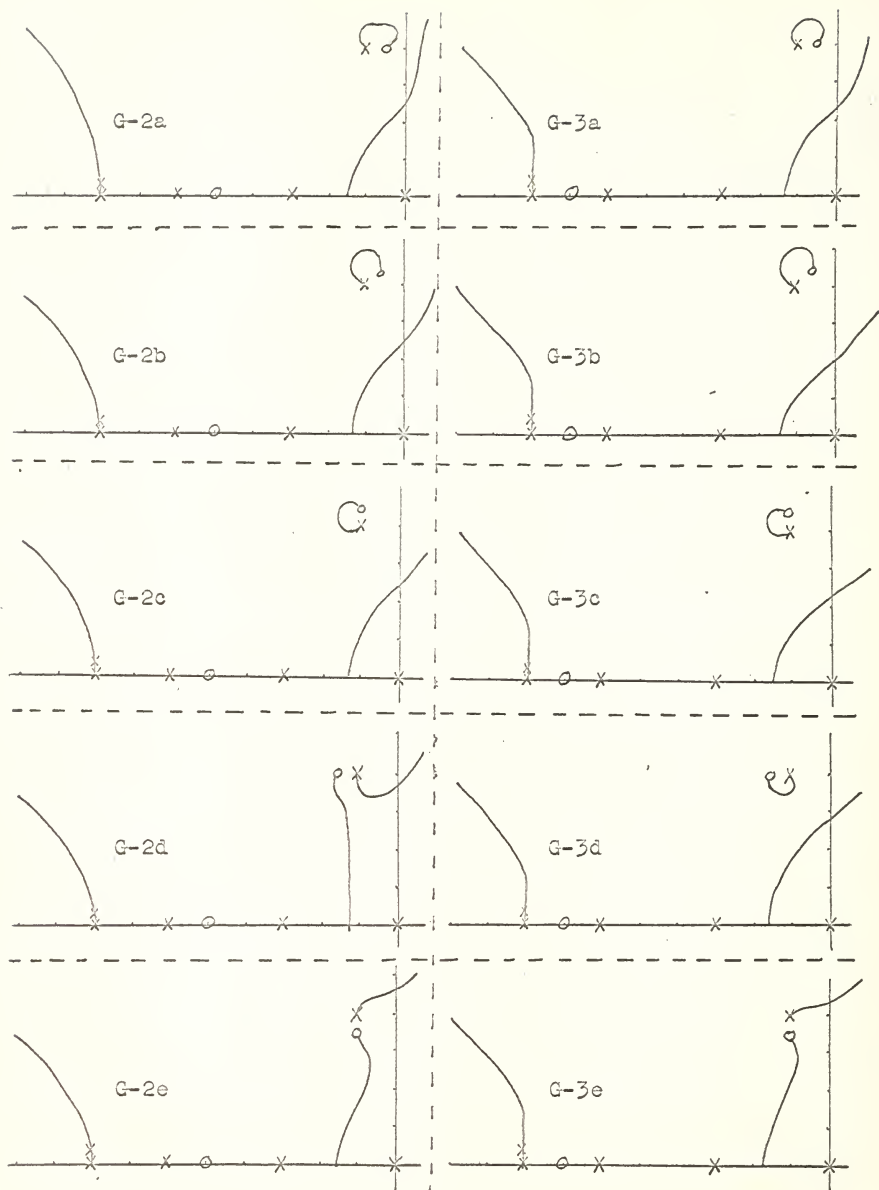


Fig. 10r Root Locus Data

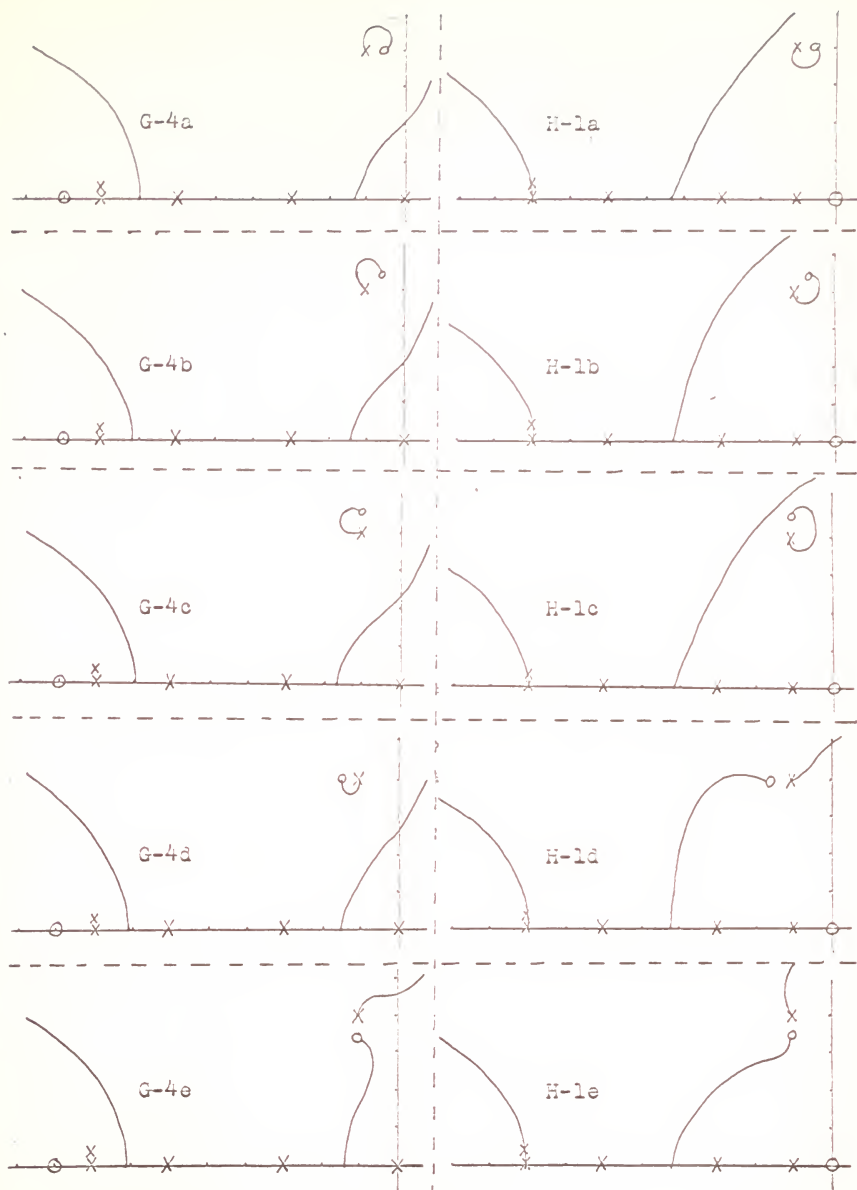


Fig. 10s Root Locus Data

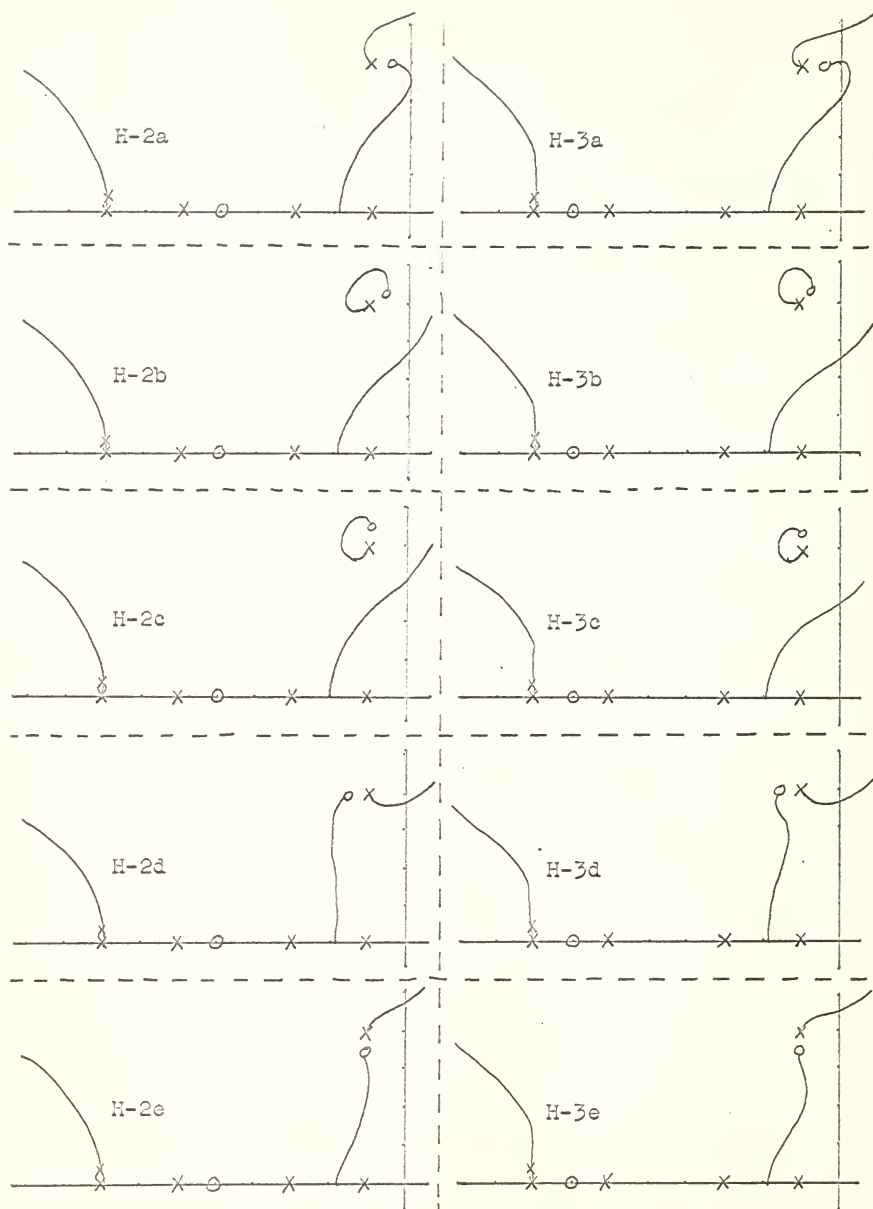


Fig. 10t Root Locus Data

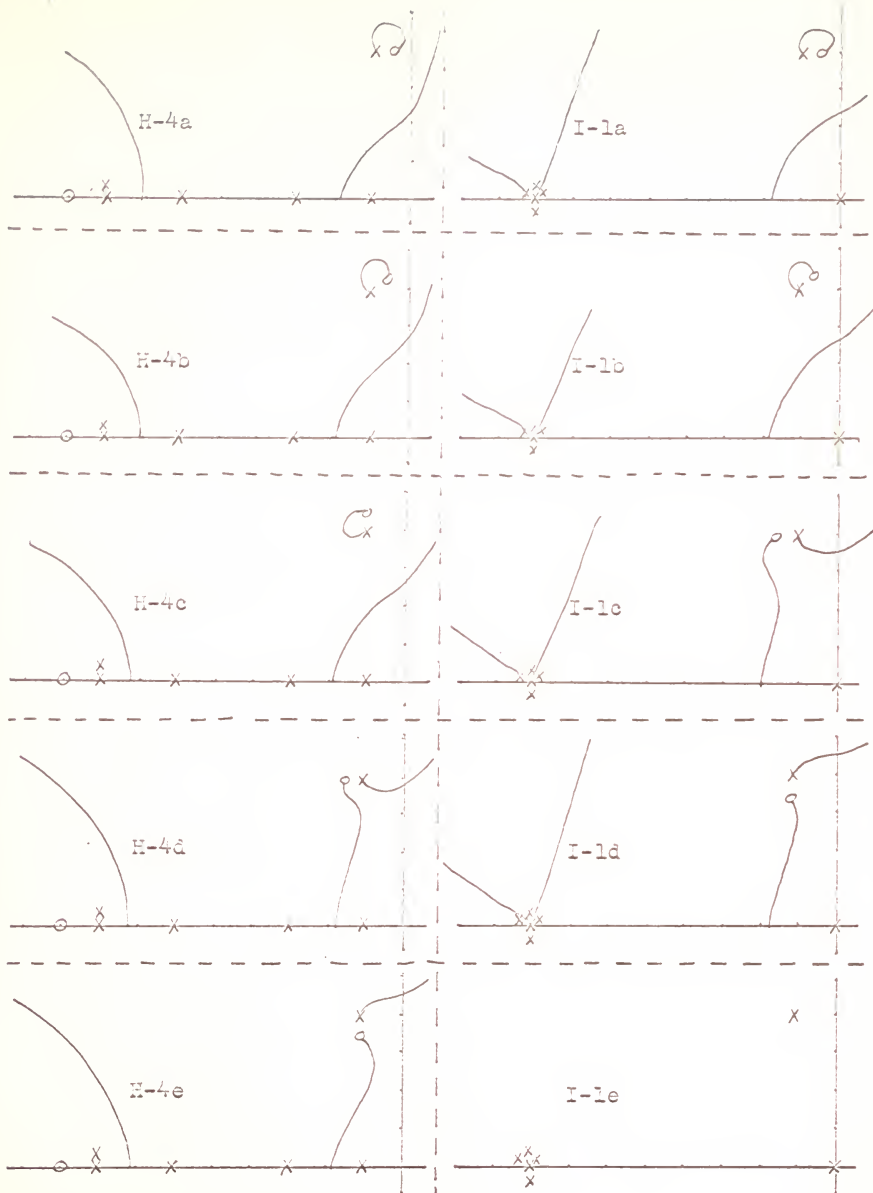


Fig. 10u Root Locus Data

2. Observations from Data

a. There is always a zero position angle θ in the first quadrant with respect to the complex pole which will cause the root locus from the complex pole to end on the near complex zero. This is observed in all cases.

b. If the root locus closes on the complex zero, then the locus from the pole moves in direction of least number of degrees (angle of emergence from pole compared with angle of emergence from zero).

c. The locus closes on complex zero if there is a root locus segment on negative real axis to the right (closest to the origin) of the real part of the complex conjugate pole. (C-1, C-2, D-1, D-2, D-3)

d. Just as the radial distance of zero had little effect on angle of emergence, it also has little effect on root locus plots except to make plots proportionately larger or smaller. (Compare Case A-1, A-5, and A-6).

e. The locus from complex pole closes on complex zero if asymptote vector is 90 degrees and the asymptote centroid is far enough out on negative real axis. (i.e., Number of pole minus number of zeros is equal to two). See cases B-1, B-2, B-3, and B-4.

TABLE 1
SUMMARY OF DATA*

Desig	θ	C, NC	Transfer Function; Other Data
A-1a	000	NC	$\frac{K(s+a-jb)(s+a+jb)}{s(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_P = 106^\circ \quad \psi_c = 196^\circ \quad \psi_z = 344^\circ$
A-1b	045	C	
A-1c	090	C	
A-1d	135	NC	
A-2a	000	NC	$\frac{K(s+a-jb)(s+a-jb)}{s(s+4)^2(s+1-j4)(s+1+j4)}$
A-2b	045	C	
A-2c	090	C	
A-2d	135	C	
A-2e	180	NC	$\angle K_P = 61^\circ \quad \psi_c = 151^\circ \quad \psi_z = 29^\circ$
A-2f	225	NC	
A-2g	315	NC	
A-3a	000	C	
A-3b	090	C	$\frac{K(s+a-jb)(s+a-jb)}{(s+1)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_P = 120^\circ \quad \psi_c = 210^\circ \quad \psi_z = 330^\circ$
A-3c	180	NC	
A-3d	270	NC	
A-4a	000	C	
A-4b	045	C	$\frac{K(s+a-jb)(s+a+jb)}{(s+2)(s+8)^2(s+1-j4)(s+1+j4)}$
A-4c	090	C	
A-4d	135	C	
A-4e	180	NC	
A-4f	225	NC	$\angle K_P = 134^\circ \quad \psi_c = 224^\circ \quad \psi_z = 316^\circ$
A-4g	270	C	
A-4h	315	C	
A-5a	000	NC	Same as A-1 except complex zeros are on .25 circle.
A-5b	045	C	
A-5c	090	C	
A-5d	135	NC	
A-5e	180	NC	
A-5f	270	NC	
A-5g	315	NC	
A-6a	000	NC	Same as A-1 except complex zeros are on 1.0 circle.
A-6b	045	C	
A-6c	090	C	
A-6d	135	NC	
A-6e	180	NC	
A-6f	225	NC	

*"Desig" is the designation of each root locus plot, θ (theta) is the angular position of the complex zero measured with respect to its adjacent complex pole, "C" indicates that the root locus from the complex pole closes or ends on its adjacent complex zero, "NC" indicates that the root locus does not close on the complex zero. Values of $\angle K_P$ were found by use of a spirule, and the values of ψ_c and ψ_z were calculated by use of the formulas developed in Chapter II with θ equal to 000 degrees.

Table 1 (Cont'd)

Design	θ	C, NC	Transfer Function; Other Data
B-1a	000	C	$\frac{K(s+1)(s+a-jb)(s+a+jb)}{s(s+8)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 196 \quad \psi_c = 286 \quad \psi_z = 254$
B-1b	045	C	
B-1c	090	C	
B-1d	180	C	
B-1e	270	C	
B-2a	000	C	$\frac{K(s+7)(s+a-jb)(s+a+jb)}{s(s+8)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 170 \quad \psi_c = 260 \quad \psi_z = 280$
B-2b	045	C	
B-2c	090	C	
B-2d	180	C	
B-2e	270	C	
B-3a	000	C	$\frac{K(s+9)(s+a-jb)(s+a+jb)}{s(s+8)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 132 \quad \psi_c = 222 \quad \psi_z = 318$
B-3b	045	C	
B-3c	090	C	
B-3d	135	C	
B-3e	180	C	
B-3f	270	C	
B-3g	315	C	
B-4a	000	C	$\frac{K(s+1)(s+a-jb)(s+a+jb)}{s(s+4)^2(s+1+j4)(s+1-j4)}$ $\underline{K_p} = 150 \quad \psi_c = 240 \quad \psi_z = 300$
B-4b	090	C	
B-4c	180	C	
B-4d	270	C	
B-5a	000	C	$\frac{K(s+3)(s+a-jb)(s+a+jb)}{s(s+4)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 124 \quad \psi_c = 214 \quad \psi_z = 326$
B-5b	045	C	
B-5c	090	C	
B-5d	225	NC	
B-5e	270	NC	
B-6a	000	NC	$\frac{K(s+5)(s+a+jb)(s+a-jb)}{s(s+4)^2(s+1-j4)(s+1+j4)}$
B-6b	045	C	
B-6c	090	C	
B-6d	180	NC	
B-6e	270	NC	
C-1a	000	C	$\frac{K(s+a+jb)(s+a-jb)}{s(s+5)(s+8)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 10 \quad \psi_c = 100 \quad \psi_z = 80$
C-1b	045	C	
C-1c	090	C	
C-1d	135	C	
C-1e	180	C	
C-1f	225	C	
C-1g	270	C	
C-1h	315	C	
C-2a	000	C	
C-2b	045	C	
C-2c	090	C	

Table 1 (Cont'd)

Desig	θ	C, NC	Transfer Function; Other Data
C-2d	135	C	$\frac{K(s+a+jb)(s+a-jb)}{s(s+1)(s+8)^2(s+1+j4)(s+1-j4)}$ $\angle K_p = 18 \quad \psi_c = 108 \quad \psi_z = 72$
C-2e	180	C	
C-2f	225	C	
C-2g	270	C	
C-2h	315	C	
C-3a	000	C	$\frac{K(s+a+jb)(s+a-jb)}{s(s+2)(s+8)^2(s+1+j4)(s+1-j4)}$ $\angle K_p = 30 \quad \psi_c = 120 \quad \psi_z = 60$
C-3b	045	C	
C-3c	090	C	
C-3d	135	C	
C-3e	180	C	
C-3f	225	C	
C-3g	270	NC	
C-3h	315	NC	
C-4a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{s(s+6)(s+8)^2(s+1+j4)(s+1-j4)}$ $\angle K_p = 68 \quad \psi_c = 158 \quad \psi_z = 22$
C-4b	045	C	
C-4c	090	C	
C-4d	135	NC	
C-4e	225	NC	
C-4f	315	NC	
C-5a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{s(s+8)^3(s+1-j4)(s+1+j4)}$ $\angle K_p = 75 \quad \psi_c = 165 \quad \psi_z = 15$
C-5b	045	C	
C-5c	090	C	
C-5d	135	NC	
C-5e	180	NC	
C-5f	225	NC	
C-5g	270	NC	
C-5h	315	NC	
C-6a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{(s+1)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 82 \quad \psi_c = 172 \quad \psi_z = 8$
C-6b	045	C	
C-6c	090	NC	
C-6d	135	NC	
C-6e	225	NC	
C-7a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{(s+2)(s+6)(s+8)^2(s+1+j4)(s+1-j4)}$ $\angle K_p = 95 \quad \psi_c = 185 \quad \psi_z = 35.5$
C-7b	045	C	
C-7c	090	NC	
C-8a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{(s+8)^4(s+1+j4)(s+1-j4)}$ $\angle K_p = 145 \quad \psi_c = 239 \quad \psi_z = 301$
C-8b	045	C	
C-8c	090	C	
C-8d	135	C	
C-8e	225	C	
C-8f	270	C	
C-8g	315	C	

Table 1 (Cont'd)

Design	θ	C, NC	Transfer Function; Other Data
D-1a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{s(s+2)(s+8)^2(s+3+j4)(s+3-j4)}$ $\underline{K_p} = 321 \quad \psi_c = 51 \quad \psi_z = 129$
D-1b	045	C	
D-1c	090	C	
D-1d	135	C	
D-1e	225	C	
D-1f	315	C	
D-2a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{s(s+3)(s+8)^2(s+3-j4)(s+3+j4)}$ $\underline{K_p} = 333 \quad \psi_c = 63 \quad \psi_z = 117$
D-2b	045	C	
D-2c	135	C	
D-2d	225	C	
D-2e	315	C	
D-2f			
D-3a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{(s+3)^2(s+8)^2(s+3+j4)(s+3-j4)}$ $\underline{K_p} = 12 \quad \psi_c = 102 \quad \psi_z = 78$
D-3b	045	C	
D-3c	090	C	
D-3d	135	C	
D-3e	225	C	
D-3f	315	C	
D-4a	000	C	$\frac{K(s+a+jb)(s+a-jb)}{s(s+8)^2(s+10)(s+3+j4)(s+3-j4)}$ $\underline{K_p} = 36 \quad \psi_c = 126 \quad \psi_z = 54$
D-4b	045	C	
D-4c	090	C	
D-4d	135	C	
D-4e	180	C	
D-4f	225	C	
D-4g	270	NC	$\underline{K_p} = 36 \quad \psi_c = 126 \quad \psi_z = 54$
D-4h	315	C	
D-4i			
D-4j			
D-4k			
D-4l			
D-5a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{(s+2)(s+6)(s+8)^2(s+3-j4)(s+3+j4)}$ $\underline{K_p} = 36 \quad \psi_c = 126 \quad \psi_z = 54$
D-5b	045	C	
D-5c	090	C	
D-5d	135	C	
D-5e	225	NC	
D-5f	315	NC	
D-6a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{(s+2)(s+8)^2(s+10)(s+3+j4)(s+3-j4)}$ $\underline{K_p} = 58 \quad \psi_c = 148 \quad \psi_z = 32$
D-6b	045	C	
D-6c	090	C	
D-6d	135	NC	
D-6e	225	NC	
D-6f	315	NC	
E-1a	000	C	$\frac{K(s+3)(s+a-jb)(s+a+jb)}{s(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\underline{K_p} = 132 \quad \psi_c = 222 \quad \psi_z = 318$
E-1b	045	C	
E-1c	090	C	
E-1d	225	NC	
E-1e	270	NC	
E-1f			

Table 1 (Cont'd)

Design	θ	C, NC	Transfer Function; Other Data
E-2a	000	C	$\frac{K(s+5)(s+a-jb)(s+a+jb)}{s(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 113 \quad \psi_c = 203 \quad \psi_z = 337$
E-2b	045	C	
E-2c	090	C	
E-2d	180	NC	
E-2e	270	NC	
E-3a	000	NC	$\frac{K(s+7)(s+a-jb)(s+a+jb)}{s(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 101 \quad \psi_c = 191 \quad \psi_z = 349$
E-3b	045	C	
E-3c	090	NC	
E-3d	180	NC	
E-3e	270	NC	
E-4a	000	NC	$\frac{K(s+9)(s+a-jb)(s+a+jb)}{s(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 95 \quad \psi_c = 185 \quad \psi_z = 355$
E-4b	045	C	
E-4c	090	NC	
E-4d	180	NC	
E-4e	270	NC	
F-1a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{s(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 1 \quad \psi_c = 91 \quad \psi_z = 89$
F-1b	045	C	
F-1c	090	C	
F-1d	180	C	
F-1e	225	NC	
F-1f	315	NC	
F-2a	000	NC	$\frac{K(s+a-jb)(s+a+jb)}{s(s+6)^2(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 27 \quad \psi_c = 117 \quad \psi_z = 63$
F-2b	045	C	
F-2c	090	C	
F-2d	180	NC	
F-2e	225	NC	
F-2f	315	NC	
F-3a	000	NC	$\frac{K(s+a+jb)(s+a-jb)}{s(s+8)^2(s+10)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 58 \quad \psi_c = 148 \quad \psi_z = 32$
F-3b	045	C	
F-3c	090	C	
F-3d	180	NC	
F-3e	315	NC	
F-4a	000	C	$\frac{K(s+a-jb)(s+a+jb)}{(s+1)(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 20 \quad \psi_c = 110 \quad \psi_z = 70$
F-4b	045	C	
F-4c	090	C	
F-4d	225	NC	
F-4e	315	NC	
G-1a	000	NC	$\frac{K(s+1)(s+a-jb)(s+a+jb)}{s(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$ $\angle K_p = 95 \quad \psi_c = 185 \quad \psi_z = 355$
G-1b	045	C	
G-1c	090	NC	
G-1d	180	NC	
G-1e	270	NC	

Table 1 (Cont'd)

Design	θ	C, NC	Transfer Function; Other Data
G-2a	000	C	$K(s+5)(s+a-jb)(s+a+jb)$
G-2b	045	C	$\frac{K(s+5)(s+a-jb)(s+a+jb)}{s(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
G-2c	090	C	
G-2d	180	NC	
G-2e	270	NC	$\underline{K_p} = 50 \quad \psi_c = 140 \quad \psi_z = 40$
G-3a	000	C	$K(s+a-jb)(s+a+jb)(s+7)$
G-3b	045	C	$\frac{K(s+a-jb)(s+a+jb)(s+7)}{s(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
G-3c	090	C	
G-3d	180	C	
G-3e	270	NC	$\underline{K_p} = 36 \quad \psi_c = 126 \quad \psi_z = 54$
G-4a	000	C	$K(s+9)(s+a-jb)(s+a+jb)$
G-4b	045	C	$\frac{K(s+9)(s+a-jb)(s+a+jb)}{s(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
G-4c	090	C	
G-4d	180	C	
G-4e	270	NC	$\underline{K_p} = 30 \quad \psi_c = 120 \quad \psi_z = 60$
H-1a	000	C	$Ks(s+a-jb)(s+a+jb)$
H-1b	045	C	$\frac{Ks(s+a-jb)(s+a+jb)}{(s+1)(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
H-1c	090	C	
H-1d	180	NC	
H-1e	270	NC	$\underline{K_p} = 123 \quad \psi_c = 213 \quad \psi_z = 327$
H-2a	000	NC	$K(s+5)(s+a-jb)(s+a+jb)$
H-2b	045	C	$\frac{K(s+5)(s+a-jb)(s+a+jb)}{(s+1)(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
H-2c	090	C	
H-2d	180	NC	
H-2e	270	NC	$\underline{K_p} = 60 \quad \psi_c = 150 \quad \psi_z = 30$
H-3a	000	NC	$K(s+7)(s+a-jb)(s+a+jb)$
H-3b	045	C	$\frac{K(s+7)(s+a-jb)(s+a+jb)}{(s+1)(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
H-3c	090	C	
H-3d	180	NC	
H-3e	270	NC	$\underline{K_p} = 52 \quad \psi_c = 142 \quad \psi_z = 38$
H-4a	000	NC	$K(s+9)(s+a-jb)(s+a+jb)$
H-4b	045	C	$\frac{K(s+9)(s+a-jb)(s+a+jb)}{(s+1)(s+3)(s+6)(s+8)^2(s+1-j4)(s+1+j4)}$
H-4c	090	C	
H-4d	180	NC	
H-4e	270	NC	
I-1a	000	C	$K(s+a-jb)(s+a+jb)$
I-1b	045	C	$\frac{K(s+a-jb)(s+a+jb)}{s(s+10)^5(s+1+j4)(s+1-j4)}$
I-1c	180	NC	
I-1d	270	NC	$\underline{K_p} = 15 \quad \psi_c = 105 \quad \psi_z = 75$

CHAPTER IV

CONCLUSIONS

Two analytical methods for obtaining the angle of emergence from a complex pole or zero have been presented. In the first of these methods, the origin was transferred to the complex pole or zero (w-plane) and then the angle of emergence was defined by equations (2-2a) and (2-3a).

The second method was to determine the residue angle by means of Heaviside's expansion theorem and then simply add 180 degrees to the residue angle to obtain the angle of emergence.

The following equations are exact, i.e., no approximations were used.

$$\psi'_c = \angle K_p + 180 \quad (2-5)$$

$$\psi'_z = \angle K_z + 180 \quad (2-7a)$$

The easiest method to obtain the angle of emergence is to find the residue angle with a spirule, then add 180 degrees.

Other relations developed in Chapter 2 are:

$$\psi_c = \theta + \psi_p - 90 \quad (2-13)$$

$$\psi_z = -\psi_c + 2\theta + 180 \quad (2-19)$$

$$\psi_z = \theta - \angle K_p + 90 \quad (2-21)$$

$$\psi_z - \psi_c = -2 \angle K_p \quad (2-22)$$

and since

$$\psi_p = \angle K_p - 180$$

$$\psi_c = \theta + \angle K_p + 90 \quad (3-1)$$

It must be remembered that these equations are only approximate with the linearizing assumptions and approximations which have been made. Figures 7 and 8 are computer solutions of the angle of emergence vs. radial position of complex zero and the angle of emergence vs. zero position angle respectively, and thus verifies the linear approximations for the assumed separation distance between complex conjugate poles.

Thus knowing the angle of emergence from a complex pole ψ_c , the position angle θ of the neighboring complex zero, the angle of emergence from the complex zero is readily obtained by equation (2-19). Similarly if the angle of emergence from a complex pole of an uncompensated system is known, the effect of adding complex zeros on the angle of emergence is obtained by using equation (2-13). Equation (2-22) reveals that the difference between the angles of emergence from complex pole and zero is proportional to the "uncompensated" residue angle and is independent of zero position angle θ .

The angle of emergence from complex pole and zero help to predict the manner in which the loci leave the pole or zero. A vector is drawn from the pole in direction of the angle of emergence and a second vector is drawn from the zero in the direction of its angle of emergence. Then the tails of these vectors are extended until they intersect at a point and thus defines an angle less than or equal to 180 degrees. It was observed that whenever the root locus from a pole closes on its adjacent zero, the loci moves in such a direction so as to be contained within the angle defined by the angle of emergence vectors.

It appears that there is always a zero position angle between 0 and 90 degrees (the first quadrant with respect to the complex pole) such that the root locus from the complex pole closes on the complex zero. Thus a self-adaptive zero compensator could be located within this 90 degree sector and effectively "compensate" the poles. The residues from the roots of these closed root loci would be negligible compared with the dominant roots of the control system.

BIBLIOGRAPHY

1. Thaler, G. J. and R. G. Brown. Analysis and Design of Feedback Control Systems. McGraw-Hill, 1960.
2. Truxal, J. G. Automatic Feedback Control System Synthesis. McGraw-Hill, 1955.
3. McCamey, R. E. and G. J. Thaler, "Design of Some Active Compensators of Feedback Controls." Applications and Industry, May, 1963.

APPENDIX I

Two digital computer programs were used. The first was "Program ANGMERG" to determine the angle of emergence for various pole and zero locations. Figures 7 and 8 were outputs of this program which verified analytical developments of Chapter II.

The second digital computer program was "Program Eqn" which is a program designed to plot root locus for any given characteristic equation (up to order 30). This program was originally programmed by LT Joseph D. Fennick, Jr., USN. A few changes in the original subroutine were made to facilitate data input, reduce printout, and speed up graph plotting time. Figures 11, 12, and 13 are sample outputs of this program. Table 2 contains all of the input data (characteristic equations) for each of the 215 individual cases studied.

..JOB0414F, REES, E.G.

PROGRAM ANGMERG

```
C THIS PROGRAM CALCULATES THE ANGLE OF EMERGENCE FOR VARIOUS
C POLE AND ZERO LOCATIONS. MULTIPLE RUNS CAN BE MADE BY
C INDICATING NUMBER OF RUNS AND ADDING ADDITIONAL DATA CARDS
C REAL AND IMAGINARY PARTS OF POLES AND ZEROS ARE READ IN
C WITH STIPULATION THAT REP(1) AND YMP(1) BE COMPLEX POLE ABOUT
C WHICH THE ANGLE OF EMERGENCE IS CALCULATED. COMPENSATION
C ZEROS ARE NOT READ IN, BUT ARE GENERATED IN THE PROGRAM.
C LET REP(2) AND YMP(2) BE CONJUGATE OF REP(1) AND YMP(1)
C PART OF THE PRINT OUT AND GRAPHS CAN BE SUPPRESSED BY
C CHANGING VALUES OF W AND V RESPECTIVELY. FOR A CURVE OF
C RADIUS VS ANGLE OF EMERGENCE AT A PARTICULAR ANGLE OF
C COMPENSATION ZERO FROM COMPLEX POLE, ANGLE AAG IS READ IN
C IN RADIANS. TWO GRAPHS OF ANGLE OF EMERGENCE VS ANGLE
C POSITION OF COMPENSATING ZEROS AT A GIVEN RADIUS IS AVAILABLE.
C RR1 AND RR2 ARE THE RADII FOR THESE CURVES. ADDITIONAL
C SYMBOLS ARE-- CE TROID, THE INTERSECTION OF ASYMPTOTES,,
C THETA -- ANGLE OF ASYMPTOTES FROM HORIZONTAL WHEN COM PENSATION
C ZEROS ARE ADDED,, ZETA - ANGLE OF ASYMPTOTES FROM HORIZONTAL
C WITHOUT COMPENSATING ZEROS,, ANGZ - SUMMATION OF ANGLES TO
C ALL ZEROS FROM COMPLEX POLE,, ANGP - SUMMATION OF ANGLES TO
C ALL POLES FROM COMPLEX POLE.
C DATA READ IN AS FOLLOWS.....
C NOS 12 NUMBER OF RUNS
C M,N 212 NO. OF POLFS M, NO. OF NON. COMP. ZEROS N
C REP(I) 8F10.3 REAL PART OF POLES
C YMP(I) 8F10.3 IMAG. PART OF POLES
C REZ(I) 8F10.3 REAL PART OF ZEROS
C YMZ(I) 8F10.3 IMAG. PART OF ZEROS
C RR1,RR2, AAG, W, V 5F10.3 RR1 AND RR2 DESIRED RADIUS FOR
C GRAPH, AAG IS ANGLE IN RADIANS LOCATING POSITION OF
C COMPENSATING ZERO FOR RADIUS VS ANGLE OF EMERGENCE GRAPH
C W SUPPRESSES PRINTOUT AS FOLLOWS (W = 0.0 PRINTS ONLY EIGHT
C VALUES, W = 1.0 PRINTS OUT SIXTEEN VALUES FOR EACH RADIUS)
C V SUPPRESSES GRAPHS AS FOLLOWS ( V = 0.0 ALL GRAPHS
C SUPPRESSED, V = 1.0 A TOTAL OF THREE GRAPHS PER RUN
C DIMENSION REP(8),YMP(8),REZ(8),YMZ(8),R(900),B(900),REZ1(900) ,
C 1ZANG(900),ANG(900),ITITLE(12),YMZ1(900),ZA(8),PA(8),GNA(900),
C 2 D(900), AZ(900), CEN(900), ACE(900)
C READ 63, NOS
63 FORMAT(I2)
C DO 402 K= 1,NOS
C READ 2,M,N
C 2 FORMAT (2I2)
C READ 4, (REP(I),I = 1,M)
C READ 4, (YMP(I),I= 1,M)
C IF(N-1) 4,9,9
C 9 READ 4, (REZ(I),I=1,N)
C READ 4, (YMZ(I), I= 1,N)
C 4 FORMAT (8F10.3)
C READ 44, RR1, RR2, AAG,W,V
C 44 FORMAT (5F10.3)
C PRINT DATA READ IN
C PRINT 62,K
```

```

62 FORMAT (/4X,7HRUN NO.,12)
   PRINT 6,M,N
6   FORMAT(/9X,2HM=,13,20X,2HN=,13)
   PRINT 8
8   FORMAT (/5X,7H REP(I),9X,7H YMP(I),9X,7H REZ(I),9X,7H YMZ(I))
   IF(N-1) 48,49,49
48  PRINT 99, (REP(I),YMP(I), I= 1,M)
   GO TO 99
49  PRINT 10, (REP(I),YMP(I),REZ(I),YMZ(I),I=1,N)
   LL = N + 1
   PRINT 99, (REP(I), YMP(I), I=LL,M)
99  FORMAT (2F15.3)
10  FORMAT (4F15.3)
   PRINT 45, RR1,RR2,AAG,W,V
45  FORMAT (/4X,5HRR1 =,F10.3, 5X,5HRR2 =,F10.3, 5X,5HAAG =,F10.3,
15X, 3HW =,F10.3,5X,3HV =,F10.3)
   XX = M-N
   X = XX - 2.
C   ASYMPTOTE VECTOR WITH COMPENSATING ZEROS, THETA
   IF (X) 202,202,203
202 PRINT 205
205  FORMAT(/2X,15HTHETA= INFINITY)
   GO TO 206
203  THETA = 180./X
   PRINT 204, THETA
204  FORMAT (/2X,6HTHETA=, F10.3)
C   ASYPTOTE VECTOR WITHOUT COMPENSATING ZEROS, ZETA
206  IF (M-N) 210, 210, 211
210  PRINT 212
212  FORMAT (/2X,14HZETA= INFINITY)
   GO TO 240
211  ZETA = 180./XX
   PRINT 214, ZETA
214  FORMAT (/2X,5HZETA=,F10.3)
C   ASYMPTOTE CFNTROID, CEN = (SUM -POLES - SUM ZEROS)/ M-N
240  SUM=0.0
   DO 201 I=1,8
201  SUM=SUM + REP(I)-REZ(I)
   SUM=-SUM
C   ANGLES DUE TO NON-COMP. ZEROS, ANGZ
   ANGZ=0.0
   IF (N) 250,250, 251
251  DO 12 I = 1,N
   IF(YMZ(I)-YMP(I))71,72,73
71  IF(REZ(I)-RFP(I))74,76,77
74  ZA(I)=ATANF((YMP(I)-YMZ(I))/(RFP(I)-REZ(I)))+3.14159
   GO TO 12
72  IF(REZ(I)-REP(I))75,80,80
75  ZA(I)=3.14159
   GO TO 12
76  ZA(I)=3./2.*3.14159
   GO TO 12
77  ZA(I)=2.*3.14159-ATANF((YMP(I)-YMZ(I))/(REZ(I)-REP(I)))
   GO TO 12
80  ZA(I)=0.0
   GO TO 12

```

```

73 IF(REZ(I)-REP(1))81,82,83
81 ZA(I)=ATANF((YMZ(I)-YMP(1))/(RFZ(I)-REP(1)))
GO TO 12
82 ZA(I)=3.14159/2.
GO TO 12
83 ZA(I) = ATANF((YMZ(I)-YMP(1))/(REZ(I) - REP(1)))
12 ANGZ=ANGZ+ZA(I)
C ANGLES DUE TO POLES, ANGP
250 ANPG = 0.0
IF (M-2) 254, 254, 253
253 DO 14 I = 3,M
IF(YMP(I)-YMP(1))91,92,93
91 IF(REP(I)-REP(1))94,96,97
94 PA(I)=ATANF((YMP(1)-YMP(I))/(REP(1)-REP(I)))+3.14159
GO TO 14
92 IF(REP(I)-REP(1))95,100,100
95 PA(I)=3.14159
GO TO 14
96 PA(I)=3./2.*3.14159
GO TO 14
97 PA(I)=2.*3.14159-ATANF((YMP(1)-YMP(I))/(REP(I)-REP(1)))
GO TO 14
100 PA(I)=0.0
GO TO 14
93 IF(REP(I)-REP(1))101,102,103
101 PA(I) = 3.14159 - ATANF((YMP(I)-YMP(1))/(REP(1) - REP(I)))
GO TO 14
102 PA(I)=3.14159/2.
GO TO 14
103 PA(I) = ATANF((YMP(I) -YMP(1))/(REP(I)-REP(1)))
14 ANPG = ANPG + PA(I)
ANGP = ANPG + 3.14159/2.
GO TO 252
254 ANGP = 3.14159/2.
C ANGLE OF EMERG. IF COMPLEX ZEROS NOT PRESENT, ANGLE
252 ANGLE=180./3.14159*(-3.14159+ANGP-ANGZ)
C GENERATE COMPENSATION ZEROS
C ALT. STATE. 15-19 FOR VARIABLE RADIUS
DO 301 J=1,20
CA = J
R(J)=CA/10.
15 DO 26 I=1,630
16 AI=I-1
17 B(I)=AI/100.
18 REZ1(I)= REP(1) - COSF(B(I))*R(J)
19 YMZ1(I) = YMP(1) + SINP(B(I))*R(J)
C ANGLES DUE TO COMPENSATION ZEROS, ZANG(I)
53 IF(B(I)-3.14159) 54,54,55
54 D(I)=180. - B(I)*180./3.14159
GO TO 56
55 D(I)=540.-B(I)*180./3.14159
56 IF(REZ1(I)-REP(1))57,58,59
57 ZANG(I)=180./3.14159*(3.14159-ATANF((-YMZ1(I)-YMP(1))/
1(REP(1)-REZ1(I))))+D(I)
GO TO 20
58 ZANG(I)=90.+D(I)

```



```

GO TO 20
59 ZANG(I)=180./3.14159*(ATANF((-YMZ1(I)-YMP(1))/(REZ1(I)-REP(1))))
  + D(I)
C   ANGLE OF EMERGENCE WITH COMPENSATION, ANG(I)
20 GNA(I)=ANGLF-ZANG(I)
  IF (X) 229,229,230
230 CFN(I) = (SMM +2. * REZ1(I) )/X
229 AZ(I) = 180. / 3.14159 * (6.28 - B(I))
21 IF(GNA(I)) 22,23,23
22 ANG(I) = -GNA(I)
  IF (ANG(I)) 600,333,333
600 ANG(I) = 360. + ANG(I)
  GO TO 333
23 ANG(I) = 360. - GNA(I)
333 IF (B(I) -AAG) 26,46,26
46 ACE(J)= ANG(I)
26 CONTINUE
  IF(R(J)-RR1) 321,27,320
320 IF(R(J)-RR2)321,27,321
27 LA=4HANG
  IF(V) 31,29,31
29 GO TO 321
31 DO 1 I=1,12
  1 ITITLE(I)=8H
  ITITLE(1)=8H REFS
  ITITLE(2)=8H E.G.
  ITITLE(3)=8H ANGLF
  ITITLE(4)=8H OF
  ITITLE(5)=8H EMERG
  ITITLE(6)= 8H VS ZERO
  ITITLE(7) = 8H POSIT
  IF(R(J)-RR1) 330,331,330
331 ITITLE(8) = 8H R=.5
  GO TO 332
330 ITITLE(8) = 8H R=1.0
332 ITITLE(12)= 8H THESIS
  CALL DRAW (630,AZ,ANG,0,0,LA,ITITLE,0,0,2,1,
1 0,0, 7,9, 0, LAST)
321 PRINT 405, R(J)
405 FORMAT (/2X,8HRADIUS =, F10.3)
220 PRINT 28
28 FORMAT (8X,14H ANGLE TO ZERO,5X, 19H ANGLE OF EMERGENCE, 5X,
1 9H CENTROID)
  IF(W) 60,61,60
60 PRINT 30, AZ(1),ANG(1), CEN(1)
  PRINT 30, AZ(39) , ANG(39), CEN(39)
  PRINT 30, AZ(78) , ANG (78) , CEN(78)
  PRINT 30, AZ(117), ANG(117), CEN(117)
  PRINT 30, AZ(157), ANG (157), CEN (157)
  PRINT 30, AZ(196), ANG(196), CEN(196)
  PRINT 30, AZ(235), ANG(235), CEN(235)
  PRINT 30, AZ(276), ANG(276), CEN(276)
  PRINT 30, AZ(314), ANG(314), CEN(314)
  PRINT 30, AZ(353), ANG(353), CEN(353)
  PRINT 30, AZ (392), ANG(392), CEN(392)
  PRINT 30, AZ(431), ANG(431), CEN(431)

```

```

PRINT 30, AZ(471), ANG(471), CFN(471)
PRINT 30, AZ(510), ANG(510), CFN(510)
PRINT 30, AZ(549), ANG (549), CFN(549)
PRINT 30, AZ(588), ANG(588), CFN(588)
GO TO 30
61 PRINT 30, AZ(1),ANG(1), CEN(1)
PRINT 30, AZ(78) , ANG (78) , CEN(78)
PRINT 30, AZ(157), ANG (157), CEN (157)
PRINT 30, AZ(235), ANG(235), CFN(235)
PRINT 30, AZ(314), ANG(314), CEN(314)
PRINT 30, AZ (392), ANG(392), CEN(392)
PRINT 30, AZ(471), ANG(471), CEN(471)
PRINT 30, AZ(549), ANG (549), CEN(549)
30 FORMAT (3F20.3)
301 CONTINUE
PRINT 401, ANGZ, ANGP
401 FORMAT(/6HANGZ =,F10.3,5X,6HAMGP =,F10.3)
PRINT 32, ANGLE
32 FORMAT (/26HANGLE OF EMERG. W/O COMP =,F10.3)
C IF V = 0.0, NO GRAPH. IF V = 1.0 A GRAPH OF R VS ANG OF EMERG
IF(V) 66,65,66
65 GO TO 402
66 LA=4HANG
LA =4HANG
DO 47 I = 1,12
47 ITITLE (I)=8H
ITITLE(1)=8H REFS
ITITLE(2)=8HE.G.
ITITLE(3)= 8HRADIUS
ITITLE(4)= 8H VS
ITITLE(5) = 8H ANGLE
ITITLE(6) = 8HOF EMERG
CALL DRAW (20,R, ACE,0,0,LA,ITITLE,0,0,2,1,
10,0,7,9,0, LAST)
402 CONTINUE
67 END
END

```

```

1
4
1.0          1.0          12.0
-6.0         6.0
1.2          2.4         -3.77          1.0

```

```

..JOB0414F, REFS,E. G.
  PROGRAM EQN
  READ 240, NUMB
240 FORMAT (I2)
  DO 241 I = 1, NUMB
241 CALL SPLANE
  FND
  SUBROUTINE SPLANE
  DIMENSION R(129),X(129),IT(10),ROOTR(128),ROOTI(128),ITITLE(12),
  1A(129),B(129),ROOTJ(128),ROOTM(128),AP(129),AZ(129)
  COMMON R,VAR,NO,ROOTR,ROOTI,AP,X
206 MOD=1
  LAB=4H
  68 FORMAT(8E15.5)
  70 FORMAT (/,14HIMAGINARY PART,/)
  69 FORMAT (/,9HREAL PART,/)
200 FORMAT(6A8)
203 FORMAT(I3)
204 FORMAT(F10.5)
  READ 200,(ITITLE(I),I=1,6)
  PRINT 324
324 FORMAT(1H1,11HGRAPH TITLF,/)
  PRINT 200,(ITITLE(I),I=1,6)
  28 FORMAT (/,36HORDER OF THE CHARACTERISTIC EQUATION,/)
  PRINT 28
  READ 203,NO
  PRINT 203,NO
  N=NO+1
205 FORMAT(8E10.5)
207 FORMAT (8E12.5)
  22 FORMAT (/,53HCONSTANT PART OF THE COEFFICIENTS IN DESCENDING ORDE
  1R,/)
  PRINT 22
  READ 205,(A(K),K=1,N)
  PRINT 207,(A(K),K=1,N)
  23 FORMAT (/,53HVARIBLE PART OF THE COEFFICIENTS IN DESCENDING ORDE
  1R,/)
  PRINT 23
  READ 205,(B(K),K=1,N)
  PRINT 207,(B(K),K=1,N)
  VAR = .001
  ND = 8
  XSCALE = 2.0
  YSCALE=XSCALE
201 FORMAT (///// ,21H0THE SYSTEM POLES ARE,/)
  PRINT 201
  M=N
  DO 67 K=1,N
  AP(K)=A(M)
  67 M=M-1
  DO 432 K=1,30
  ROOTI(K)=0.
432 ROOTR(K)=0.
  NX=NO
  DO 15 K=1,31
  15 X(K)=B,B

```

```

CALL POLYRT (AP,X,NX,ROOTR,ROOTI,1.E-5)
PRINT 69
PRINT 68,(ROOTR(K),K=1,NO)
PRINT 70
PRINT 68,(ROOTI(K),K=1,NO)
CALL DRAW(NO,ROOTR,ROOTI,MOD,1,LAB,ITITLE,XSCALE,YSCALE,
11,7,2,2,8,9,0,LAST)
MOD=2
202 FORMAT (////////,21H THE SYSTEM ZEROS ARE,/)
K=1
3 IF(B(K)) 1,2,1
2 K=K+1
GO TO 3
1 NORD=N-K
IF(NORD-1) 6,4,5
4 ZFRO=-B(K+1)/B(K)
7 FORMAT (/,16H THE SYSTEM ZERO=,F15.5,/)
PRINT 7,ZERO
ROOTM(1)=ZERO
ROOTM(2)=16.*XSCALE
ROOTJ(1)=0.
ROOTJ(2)=0.
NORD=2
GO TO 11
6 PRINT 9
9 FORMAT (/,25H ALL ZEROS ARE AT INFINITY)
GO TO 8
5 NN=NORD+1
DO 10 L=1,NN
R(L)=B(K)
10 K=K+1
PRINT 202
M=NN
DO 46 K=1,NN
AZ(K)=R(M)
46 M=M-1
DO 433 K=1,30
ROOTM(K)=0.
433 ROOTJ(K)=0.
NORX=NORD
DO 16 K=1,31
16 X(K)=0.0
CALL POLYRT(AZ,X,NORX,ROOTM,ROOTJ,1.E-5)
PRINT 69
PRINT 68,(ROOTM(K),K=1,NORD)
PRINT 70
PRINT 68,(ROOTJ(K),K=1,NORD)
11 CALL DRAW(NORD,ROOTM,ROOTJ,MOD,3,LAB,ITITLE,XSCALE,YSCALE,
11,7,2,2,8,9,0,LAST)
8 CONTINUE
GO TO(31,32,33,34,35,36,37,38,39,40),ND
31 G=1.0076
GO TO 41
32 G=1.016
GO TO 41
33 G=1.0248

```

```

      GO TO 41
34  G=1.0312
      GO TO 41
35  G=1.0394
      GO TO 41
36  G=1.0483
      GO TO 41
37  G=1.0568
      GO TO 41
38  G=1.0633
      GO TO 41
39  G=1.071
      GO TO 41
40  G=1.078
41  CONTINUE
      DO 101 J=1,30
      DO 100 K=1,10
      DO 300 L=1,N
300  R(L)=A(L)+B(L)*VAR
      CALL ROOTX
      DO 71 JJ=1,NO
      IF (ABSF(ROOTI(JJ))-5.E-04) 61,61,62
61  ROOTJ(JJ)=0.
      GO TO 71
62  ROOTJ(JJ)=ROOTI(JJ)
71  CONTINUE
100  VAR=VAR*G
      IF (J-30) 101,98,98
98  MOD=3
101  CALL DRAW(NO,ROOTR,ROOTJ,MOD,2,LAB,ITITLE,XSCALE,YSCALE,
11,7,2,2,8,9,0, LAST)
      GO TO 206
      END
      SURROUTINE ROOTX
      DIMENSION C(31),D(29),R(129),ROOTR(128),ROOTI(128),EE(31),FE(31)
1  ,AP(129),X(129)
      COMMON R,VAR,NO,ROOTR,ROOTI,AP,X
      M=1
      DO 1492 MA=1,31
      EE(MA)=0.
1492  FF(MA)=0.
20  BETAN=ROOTI(M)+EE(M)
      ALFAN=ROOTR(M)+FE(M)
      DO 7 I=1,100
      S=2.*ALFAN
      T=-(ALFAN**2+BETAN**2)
      C(1)=R(1)
      C(2)=R(2)+S*R(1)
      NC=NO+1
      DO 2 L=2,NC
2  C(L) = R(L)+S*C(L-1) + T*C(L-2)
      AN= C(NO+1)-ALFAN*C(NO)
      BN= BETAN*C(NO)
      IF (NO-3) 21, 17, 18
17  CN = 3.*R(1)*(ALFAN**2-BETAN**2) + 2.*R(2)*ALFAN + R(3)
      DN = 6.*R(1)*ALFAN*BETAN + 2.*R(2)*BETAN

```

```

GO TO 19
21 CN = 2.*R(1)*ALFAN + R(2)
DN = 2.*R(1)*BETAN
GO TO 19
18 D(1) = C(1)
D(2)=C(2)+S*D(1)
NU=NO-1
DO 3 N=3,NU
3 D(N)=C(N)+S*D(N-1)+T*D(N-2)
CN= C(NO)-2.*D(NO-2)*BETAN**2
DN= 2.*BETAN*(D(NO-1)-ALFAN*D(NO-2))
19 ALFA=ALFAN-(AN*CN+BN*DN)/(CN**2+DN**2)
BETA =BETAN + (AN*DN-BN*CN)/(CN**2+DN**2)
AZETA=.01+ABSF(BETAN)
ZETA=SQRT(.01-T)
IF(ABSF((ALFA-ALFAN)/ZETA )-.1)55,55,52
55 IF(ABSF((BETA-BETAN)/AZETA)-1.)56,56,52
56 EE(M) =(ALFA-ALFAN)*10. + BETA - ROOTI(M)
FE(M)=(BETA - BETAN)*10. +ALFA - ROOTR(M)
IF(ABSF((ALFA-ALFAN)/(ALFAN+1.))-5.E-4)4,4,5
4 IF(ABSF((BETA-BETAN)/(BETAN+1.))-5.E-4)6,6,5
5 ALFAN=ALFA
7 BETAN=BETA
52 CONTINUE
MR=NC
DO 6777 K=1,NC
AP(K)=R(MR)
6777 MR=MR-1
DO 434 K=1,30
ROOTI(K)=0.
434 ROOTR(K)=0.
NX=NO
DO 83 K=1,31
83 X(K)=0.0
CALL POLYRT (AP,X,NX,ROOTR,ROOTI,1.E-5)
GO TO 16
6 ROOTR(M)=ALFA
1 ROOTI(M)=BETA
12 IF(M-NO) 15,16,16
15 M=M+1
GO TO 20
16 RETURN
END
END

```

2
REES, E.G. THESIS

5					
1.0	19.0	131.	513.	1488.	1088.
			1.0	3.0	18.25

REES, E. G. THESIS

6					
1.0	24.	221.	978.	2888.	6528.
				1.0	1.0
					16.25

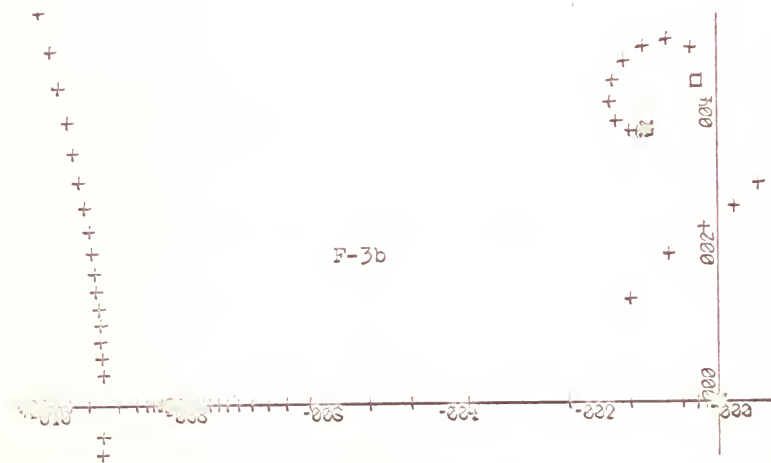
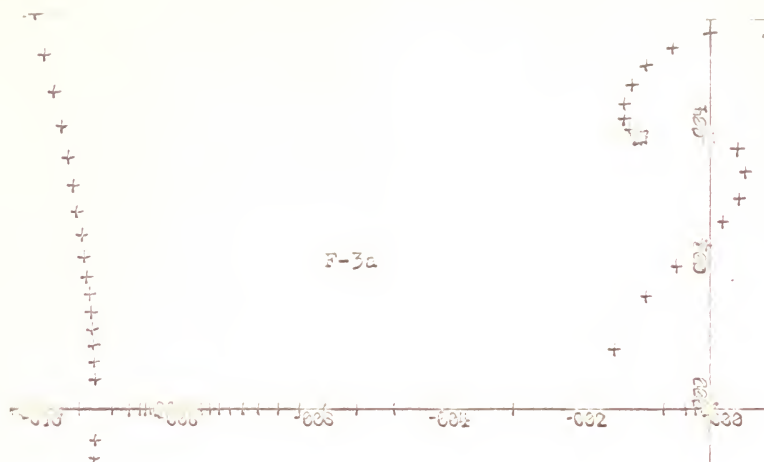


Fig. 11 Root Locus Data (Computer)

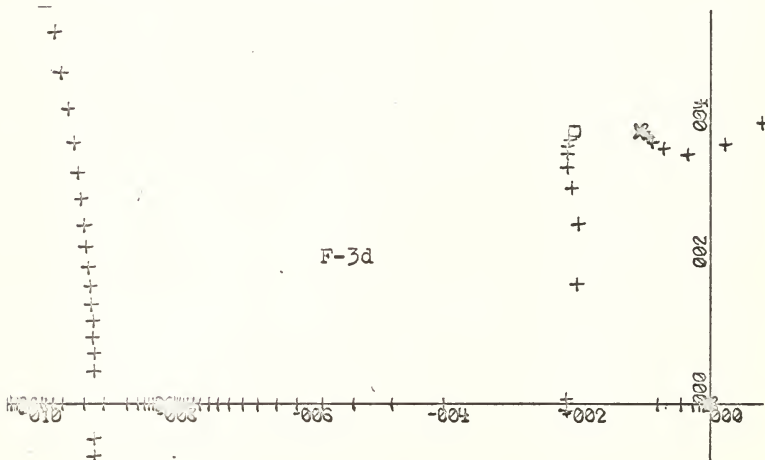
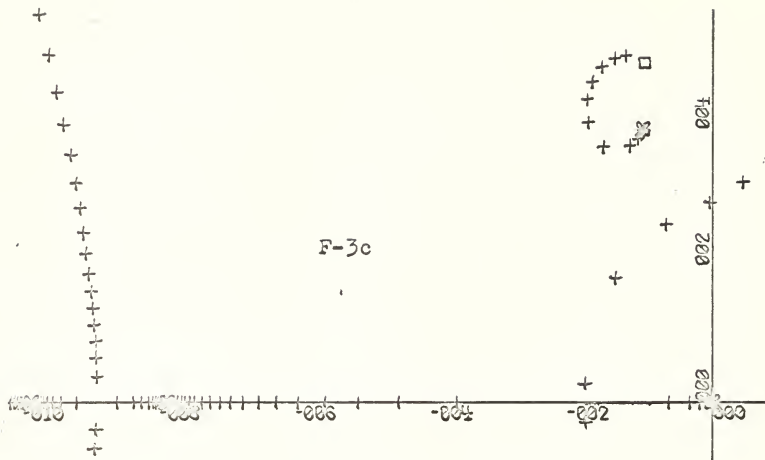


Fig. 12 Root Locus Data (Computer)

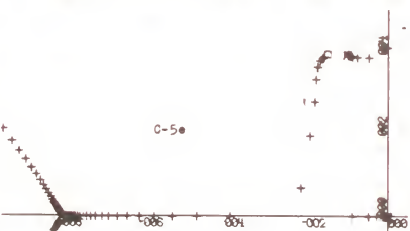
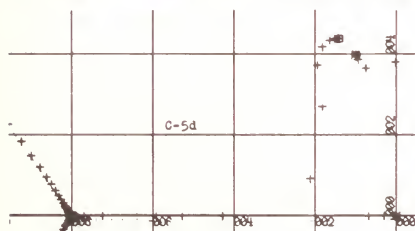
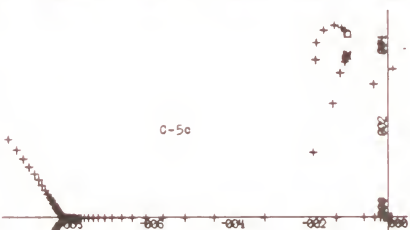
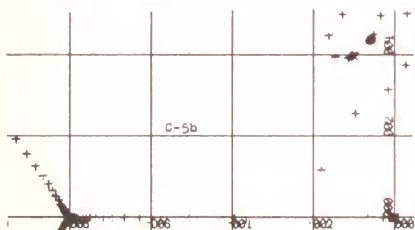
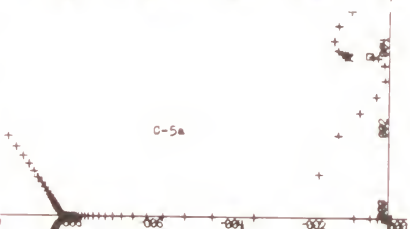
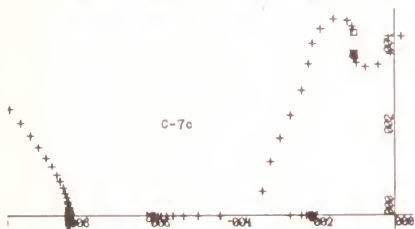
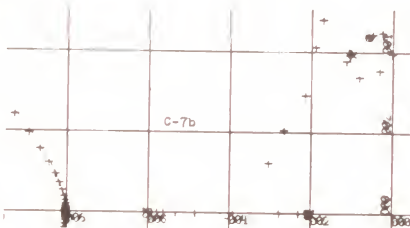
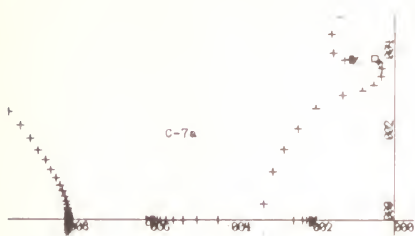


TABLE 2
CHARACTERISTIC EQUATIONS

Desig	θ	s^6	s^5	s^4	s^3	s^2	s^1	s^0
A-1	*		1	18	113	400	1088	0
A-1a	000					1	1	16.25
A-1b	045					1	1.2	19.56
A-1c	090					1	2	21.25
A-1d	135					1	2.8	21.16

A-2	*		1	10	49	168	272	0
A-2a	000					1	1	16.25
A-2b	045					1	1.2	19.56
A-2c	090					1	2	21.25
A-2d	135					1	2.8	21.16
A-2e	180					1	3	18.25
A-2f	225					1	2.8	15.06
A-2g	315					1	1.2	13.46

A-3	*		1	19	131	513	1488	1088
A-3a	000					1	1	16.25
A-3b	090					1	2	21.25
A-3c	180					1	3	18.25
A-3d	270					1	2	13.25

A-4	*		1	20	149	626	1888	2176
A-4a	000					1	1	16.25
A-4b	045					1	1.2	19.56
A-4c	090					1	2	21.25
A-4d	135					1	2.8	21.16
A-4e	180					1	3	18.25
A-4f	225					1	2.8	15.06
A-4g	270					1	2	13.25
A-4h	315					1	1.2	13.46

A-5	*		1	18	113	400	1088	0
A-5a						1	1.5	16.56
A-5b						1	1.62	18.26
A-5c						1	2.5	17.56
A-5d						1	2.38	19.01
A-5e						1	2	19.1
A-5f						1	2	15.05
A-5g						1	1.62	15.2

A-6	*		1	18	113	400	1088	0
A-6a	000					1	2	16.01
A-6b	045					1	6	22.19
A-6c	090					1	2	26.0
A-6d	135					1	34	24.97
A-6e	180					1	2	10.0

Design	0	6 s	5 s	4 s	3 s	2 s	1 s	0 s
B-1	*		1	18	113	400	1088	0
B-1a	000				1	2	17.25	16.25
B-1b	045				1	2.2	20.75	19.56
B-1c	090				1	3.0	23.25	21.25
B-1d	180				1	4.0	21.25	18.25
B-1e	270				1	3.0	15.25	13.25

B-2	*		1	18	113	400	1088	0
B-2a	000				1	8	23.25	113.75
B-2b	045				1	8.2	27.96	136.92
B-2e	090				1	9.0	35.25	148.75
B-2d	180				1	10	39.25	127.75
B-2e	270				1	9	27.25	92.75

B-3	*		1	18	113	400	1088	0
B-3a	000				1	10	25.25	146.25
B-3b	045				1	10.2	20.36	176.04
B-3c	090				1	11.0	39.25	191.25
B-3d	135				1	11.8	46.36	190.44
B-3e	180				1	12.0	45.25	164.25
B-3f	270				1	11.0	31.25	119.25
B-3g	315				1	10.2	24.26	120.94

B-4	*		1	10	49	168	272	0
B-4a	000				1	2.0	17.25	16.25
B-4b	090				1	2.2	20.75	19.56
B-4c	180				1	3.0	23.25	21.25
B-4d	270				1	4.0	21.25	18.25

B-5	*		1	10	49	168	272	0
B-5a	000				1.0	4.0	19.25	48.75
B-5b	045				1.0	4.2	23.16	58.68
B-5c	090				1.0	5.0	27.25	63.75
B-5d	225				1.0	6.0	24.25	54.75
B-5e	270				1.0	5.0	19.25	39.75

B-6	*		1	10	49	168	272	0
B-6a	000				1	6.0	21.25	81.25
B-6b	045				1	6.2	25.56	97.80
B-6c	090				1	7.0	31.25	106.25
B-6d	180				1	8.0	33.25	91.25
B-6e	270				1	7.0	23.25	66.25

C-1	*	1.0	10.5	122	456.5	1288	544	0
C-1a	000					1	1	16.25
C-1b	045					1	1.2	19.56
C-1c	090					1	2	21.25

Table 2 (Cont'd)

Desig	0	⁶ _s	⁵ _s	⁴ _s	³ _s	² _s	¹ _s	⁰ _s
C-1d	135					1	2.8	21.16
C-1e	180					1	3.0	18.25
C-1f	225					1	2.8	15.06
C-1g	270					1	2.0	13.25
C-1h	315					1	1.2	13.46

C-2	*	1	19	131	513	1488	1088	0
C-2a	000					1	1.0	16.25
C-2b	045					1	1.2	19.56
C-2c	090					1	2.0	21.25
C-2d	135					1	2.8	21.16
C-2e	180					1	3.0	18.25
C-2f	225					1	2.8	15.06
C-2g	270					1	3.0	18.25
C-2h	315					1	2	15.05

C-3	*	1	20	149	626	1888	2176	0
C-3a	000					1	1	16.25
C-3b	045					1	1.2	19.56
C-3c	090					1	2	21.25
C-3d	135					1	2.8	21.16
C-3e	180					1	3.0	18.25
C-3f	225					1	2.8	15.06
C-3g	270					1	2	13.25
C-3h	315					1	1.2	13.46

C-4	*	1	24	221	1078	3480	6528	0
C-4a	000					1	1	16.25
C-4b	045					1	1.2	19.56
C-4c	090					1	2	21.25
C-4d	135					1	2.8	21.16
C-4e	225					1	2.8	15.06
C-4f	315					1	1.2	13.46

C-5	*	1	26	257	1304	4288	8704	0
C-5a	000					1	1	16.25
C-5b	045					1	1.2	19.56
C-5c	090					1	2	21.25
C-5d	135					1	2.8	21.16
C-5e	180					1	3.0	18.25
C-5f	225					1	2.8	15.06
C-5g	270					1	2	13.25
C-5h	315					1	1.2	13.46

C-6	*	1	25	245	1299	4566	10016	0
C-6a	000					1	1	16.25
C-6b	045					1	1.2	19.56

Table 2 (Cont'd)

Design	9	6 s	5 s	4 s	3 s	2 s	1 s	0 s
C-6c	090					1	2	21.25
C-6d	135					1	2.8	21.16
C-6e	225					1	2.8	15.06
C-7	*	1	26	269	1520	5644	13504	13056
C-7a	000					1	1	16.25
C-7b	045					1	1.2	19.56
C-7c	090					1.	2	21.25
C-8	*	1	34	465	3360	14720	43008	69632
C-8a	000					1	1	16.25
C-8b	045					1	1.2	19.56
C-8c	090					1	2	21.25
C-8d	135					1	2.8	21.16
C-8e	225					1	2.8	15.06
C-8f	270					1	2	13.25
C-8g	315					1	1.2	13.46
D-1	*	1	24	229	1154	2168	3200	0
D-1a	000					1	5	22.25
D-1b	045					1	5.2	25.94
D-1c	090					1	6.0	29.25
D-1d	135					1	6.8	30.74
D-1e	225					1	6.7	24.5
D-1f	315					1	5.3	20.54
D-2	*	1	25	251	1339	3952	4800	0
D-2a	000					1	5.0	22.25
D-2b	045					1	5.2	25.94
D-2c	135					1	6.8	20.74
D-2d	225					1	6.7	24.5
D-2e	315					1	5.3	20.54
D-3	*	1	30	373	2528	10092	22208	19200
D-3a	000					1	5	22.25
D-3b	045					1	5.2	25.94
D-3c	090					1	6.0	29.25
D-3d	135					1	6.8	30.74
D-3e	225					1	6.7	24.5
D-3f	315					1	5.3	20.54
D-4	*	1	32	405	2634	9440	16000	00
D-4a	000					1	5	22.25
D-4b	045					1	5.2	25.94
D-4c	090					1	6.0	29.25
D-4d	135					1	6.8	30.74
D-4e	180					1	7.0	28.25

Table 2 (Cont'd)

Desig	0	⁶ _s	⁵ _s	⁴ _s	³ _s	² _s	¹ _s	⁰ _s
D-4f	225					1	6.7	24.5
D-4g	270					1	6.0	21.25
D-4h	315					1	5.3	20.54
D-5	*	1	30	373	2528	10092	22208	19200
D-5a	000					1	5	22.25
D-5b	045					1	5.2	25.94
D-5c	090					1	6.0	29.25
D-5d	135					1	6.8	30.74
D-5e	225					1	6.7	24.5
D-5f	315					1	5.3	20.54
D-6	*	1	34	469	3444	14708	34880	32000
D-6a	000					1	5	22.25
D-6b	045					1	5.2	25.94
D-6c	090					1	6.0	29.25
D-6d	135					1	6.8	30.74
D-6e	225					1	6.7	24.5
D-6f	315					1	5.3	20.54
E-1	*	1	24	221	1078	3488	6528	0
E-1a	000				1	4	1925	48.75
E-1b	045				1	4.2	23.16	58.68
E-1c	090				1	5.0	27.25	63.75
E-1d	225				1	6.0	24.25	54.75
E-1e	270				1	5.0	19.25	39.75
E-2	*	1	24	221	1078	3488	6528	0
E-2a	000				1	6	21.25	81.25
E-2b	045				1	6.2	25.56	97.8
E-2c	090				1	7.0	31.25	106.25
E-2d	180				1	8.0	33.25	91.25
E-2e	270				1	7.0	23.25	66.25
E-3	*	1	24	221	1078	3488	6528	0
E-3a	000				1	8	23.25	113.75
E-3b	045				1	8.2	27.96	136.92
E-3c	090				1	9.0	35.25	148.75
E-3d	180				1	10.0	39.25	217.75
E-3e	270				1	9	27.25	92.75
E-4	*	1	24	221	1078	3488	6528	0
E-4a	000				1	10	25.25	146.25
E-4b	045				1	10.2	30.36	176.04
E-4c	090				1	11.0	39.25	191.25
E-4d	180				1	12.0	45.25	164.25
E-4e	270				1	11.0	31.25	119.25

Design	0	8 _s	7 _s	6 _s	5 _s	4 _s	3 _s	2 _s	1 _s	0 _s
F-1	*		1	27	293	1741	6722	16992	19584	0
F-1a	000							1	1	16.25
F-1b	045							1	1.2	19.56
F-1c	090							1	2	21.25
F-1d	180							1	.2	16.01
F-1e	225							1	.6	22.19
F-1f	315							1	2	10
F-2	*		1	30	365	2404	9956	27456	39168	0
F-2a	000							1	1.2	19.56
F-2b	045							1	2	21.25
F-2e	090							1	.2	16.01
F-2d	180							1	.6	22.19
F-2e	225							1	2	26.0
F-2f	315							1	2	10
F-3	*		1	38	573	4460	20388	61760	108800	0
F-3a	000							1	.2	16.01
F-3b	045							1	1.62	18.26
F-3c	090							1	2	26
F-3d	180							1	2	10
F-3e	315							1	.6	10.98
F-4	*		1	28	320	2034	8463	23714	36576	19584
F-4a	000							1	1	16.25
F-4b	045							1	1.2	19.56
F-4c	090							1	2.8	21.16
F-4d	225							1	2.8	15.06
F-4e	315							1	1.2	13.46
G-1	*		1	27	293	1741	6722	16992	19584	0
G-1a	000						1	2	17.25	16.25
G-1b	045						1	2.2	20.75	19.56
G-1c	090						1	3.0	23.25	21.25
G-1d	180						1	4.0	21.25	18.25
G-1e	270						1	3.0	15.25	13.25
G-2	*		1	27	293	1741	6722	16992	19584	0
G-2a	000						1	6	21.25	81.25
G-2b	045						1	6.2	25.56	97.8
G-2c	090						1	7.0	31.25	106.25
G-2d	180						1	8.0	33.25	91.25
G-2e	270						1	7.0	23.25	66.25
G-3	*		1	27	293	1741	6722	16992	19584	0
G-3a	000						1	8	23.25	113.75
G-3b	045						1	8.2	27.96	136.92
G-3c	090						1	9.0	35.25	148.75
G-3d	180						1	10	39.25	127.75

Desig	0	s ⁸	s ⁷	s ⁶	s ⁵	s ⁴	s ³	s ²	s ¹	s ⁰
G-3e	270						1	9.0	27.25	92.75
-	-	-	-	-	-	-	-	-	-	-
G-4 *			1	27	293	1741	6722	16992	19584	0
G-4a	000						1	10	25.25	146.25
G-4b	045						1	10.2	30.36	176.04
G-4c	090						1	11	39.25	191.25
G-4d	180						1	12	45.25	164.25
G-4e	270						1	11	31.25	119.25
-	-	-	-	-	-	-	-	-	-	-
H-1 *			1	28	320	2034	8463	23714	36576	19584
H-1a	000						1	1	16.25	0
H-1b	045						1	1.2	19.56	0
H-1c	090						1	2.0	21.25	0
H-1d	180						1	3.0	18.25	0
H-1e	270						1	2.0	13.25	0
-	-	-	-	-	-	-	-	-	-	-
H-2 *			1	28	320	2034	8463	23714	36576	0
H-2a	000						1	6.0	21.25	81.25
H-2b	045						1	6.2	25.56	97.8
H-2c	090						1	7.0	31.25	106.25
H-2d	180						1	8.0	33.25	91.25
H-2e	270						1	7.0	23.25	66.25
-	-	-	-	-	-	-	-	-	-	-
H-3 *			1	28	320	2034	8463	23714	36576	19584
H-3a	000						1	8	23.25	113.75
H-3b	045						1	8.2	27.96	136.92
H-3c	090						1	9.0	35.25	148.75
H-3d	180						1	10	39.25	127.75
H-3e	270						1	9	27.25	92.75
-	-	-	-	-	-	-	-	-	-	-
H-4 *			1	28	320	2034	8463	23714	36576	19584
H-4a	000						1	10	25.25	146.25
H-4b	045						1	10.2	30.36	176.04
H-4c	090						1	11.0	39.25	191.25
H-4d	180						1	12.0	45.25	164.25
H-4e	270						1	11.0	31.25	119.25
-	-	-	-	-	-	-	-	-	-	-

INITIAL DISTRIBUTION LIST

	No. Copies
1. Defense Documentation Center Cameron Station Alexandria, Virginia 22314	20
2. Library U. S. Naval Postgraduate School Monterey, Calif.	2
3. Bureau of Ships Navy Department Washington, D. C. 20360	1
4. Prof. G. J. Thaler Department of Electrical Engineering U. S. Naval Postgraduate School Monterey, Calif.	5
5. LT E. Grant Rees, USN 109 Leidig Circle Monterey, Calif. 93940	1

DOCUMENT CONTROL DATA - R&D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1 ORIGINATING ACTIVITY (Corporate author) LT E. Grant Rees, USN U. S. Naval Postgraduate School Monterey, Calif.		2a REPORT SECURITY CLASSIFICATION Unclassified	
		2b GROUP -	
3 REPORT TITLE Some Extensions of the Root Locus Method			
4 DESCRIPTIVE NOTES (Type of report and inclusive dates) Final report			
5 AUTHOR(S) (Last name, first name, initial) E. Grant Rees, USN, LT			
6 REPORT DATE May 1966		7a TOTAL NO OF PAGES 85	7b NO OF REFS
8a CONTRACT OR GRANT NO		8a ORIGINATOR'S REPORT NUMBER(S) NA	
b. PROJECT NO.			
c. NA		8b OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
d.			
10 AVAILABILITY/LIMITATION NOTICES THIS REPORT IS UNCLASSIFIED AND UNRESTRICTED FROM DISSEMINATION BY ANY PERSON EXCEPT WHERE SHOWN OTHERWISE			
11 SUPPLEMENTARY NOTES None		12 SPONSORING MILITARY ACTIVITY U. S. Navy	
13 ABSTRACT Simple analytical methods for the determination of the angle of emergence of complex poles and zeros are developed. Several case studies of root loci emerging from complex zeros are analyzed to determine criterion for the loci starting from complex pole to end on complex zero, as a pre-requisite study for "cancel-compensation" and self-adaptive compensators.			

14. KEY WORDS	LINK A		LINK C		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Root Locus Angle of Emergence Complex residue angle Cancel Compensation						

INSTRUCTIONS

1. **ORIGINATING ACTIVITY:** Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. **REPORT SECURITY CLASSIFICATION:** Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. **GROUP:** Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. **REPORT TITLE:** Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. **DESCRIPTIVE NOTES:** If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. **AUTHOR(S):** Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. **REPORT DATE:** Enter the date of the report as day, month, year, or month, year. If more than one date appears on the report, use date of publication.

7a. **TOTAL NUMBER OF PAGES:** The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. **NUMBER OF REFERENCES:** Enter the total number of references cited in the report.

8a. **CONTRACT OR GRANT NUMBER:** If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. **PROJECT NUMBER:** Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. **ORIGINATOR'S REPORT NUMBER(S):** Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. **OTHER REPORT NUMBER(S):** If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. **AVAILABILITY/LIMITATION NOTICES:** Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. **SUPPLEMENTARY NOTES:** Use for additional explanatory notes.

12. **SPONSORING MILITARY ACTIVITY:** Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. **ABSTRACT:** Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. **KEY WORDS:** Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, roles, and weights is optional.

1000

1000

thesR274

DUDLEY KNOX LIBRARY



3 2768 00414100 2

DUDLEY KNOX LIBRARY